



TOLIMAN

Getting to know the neighbours: Earth analogues in Alpha Centauri with the TOLIMAN space telescope

Peter Tuthill
Chris Betters
Connor Langford
Karel Valenta
Fred Crous

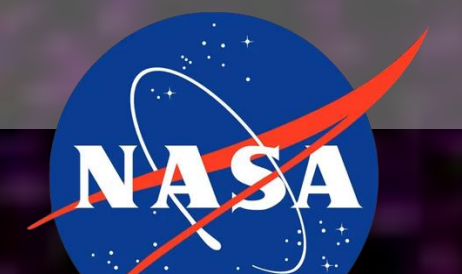
Louis Desdoigts
Max Charles

Celine Boehm

Pete Worden
Olivier Guyon
Pete Klupar
Kyran Gratan

Ben Pope
Ben Montet

Kieran Larkin
Gautam Vashisht
Frans Snik
David Doelman





TOLIMAN

Telescope for Orbit Locus Interferometric Monitoring of our Astronomical Neighborhood

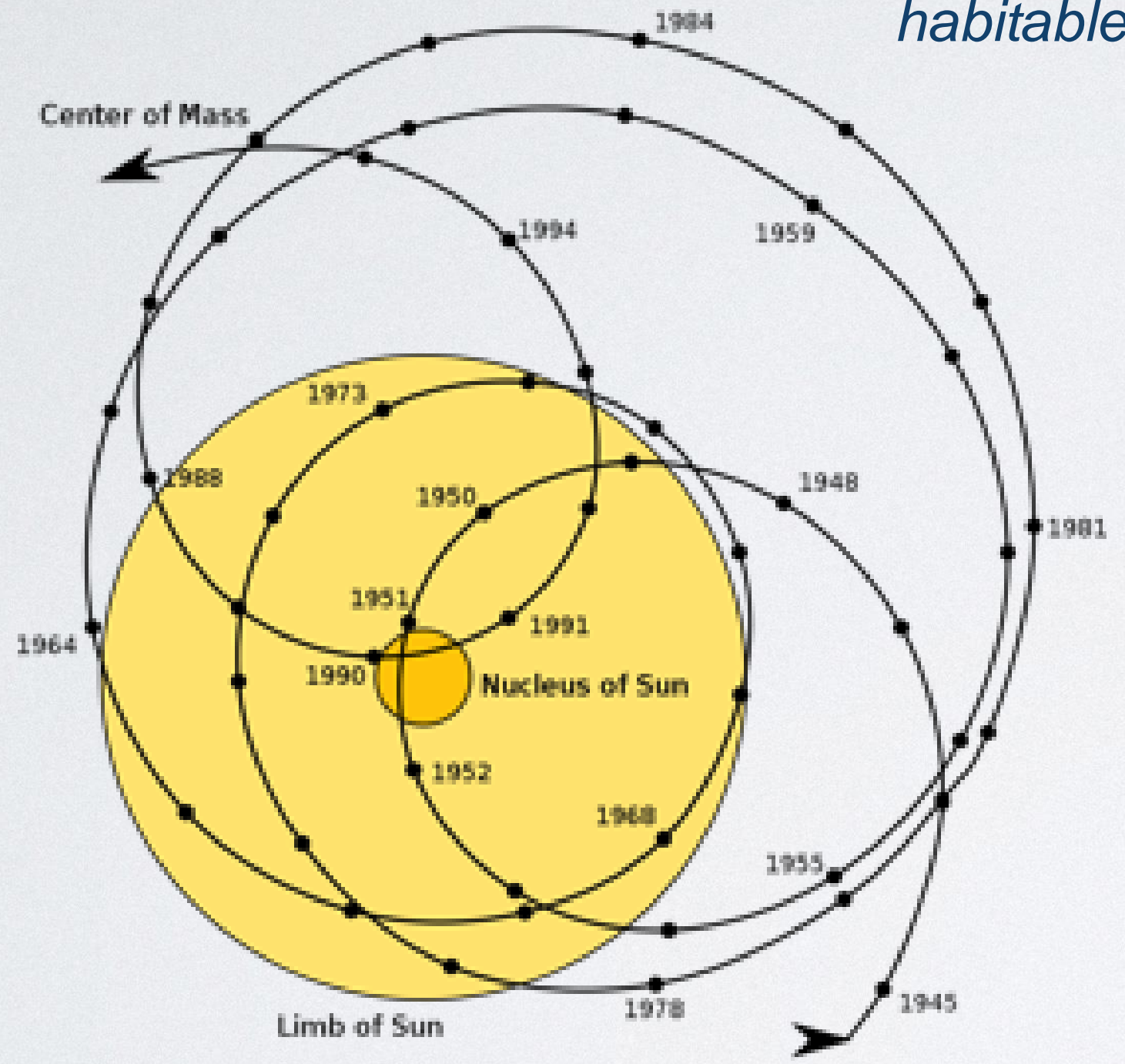
- Is there an Earth Analog in Alpha Cen?
- 1 Earth Mass, 0.5-2yr orbit, A or B
 - Secondary Target: 61 Cygni (+others?)
 - 12.5cm pathfinder for a 30cm space telescope
 - Technology demonstrator for Astrometry





How do we find rocky temperate zone planets around stars within ~ 10 Pc?

- ✗ Transits? - Sample too small to get lucky
- ✗ Doppler RV? - Signals very small for terrestrial mass temperate planet orbiting FGK star
- ✓ Astrometry? - *“Astrometry is the only technique technologically ready to detect planets of Earth mass in the habitable zone (HZ) around solar-type stars within 20 pc.”* Shao et al 2010



True Earth Analogs

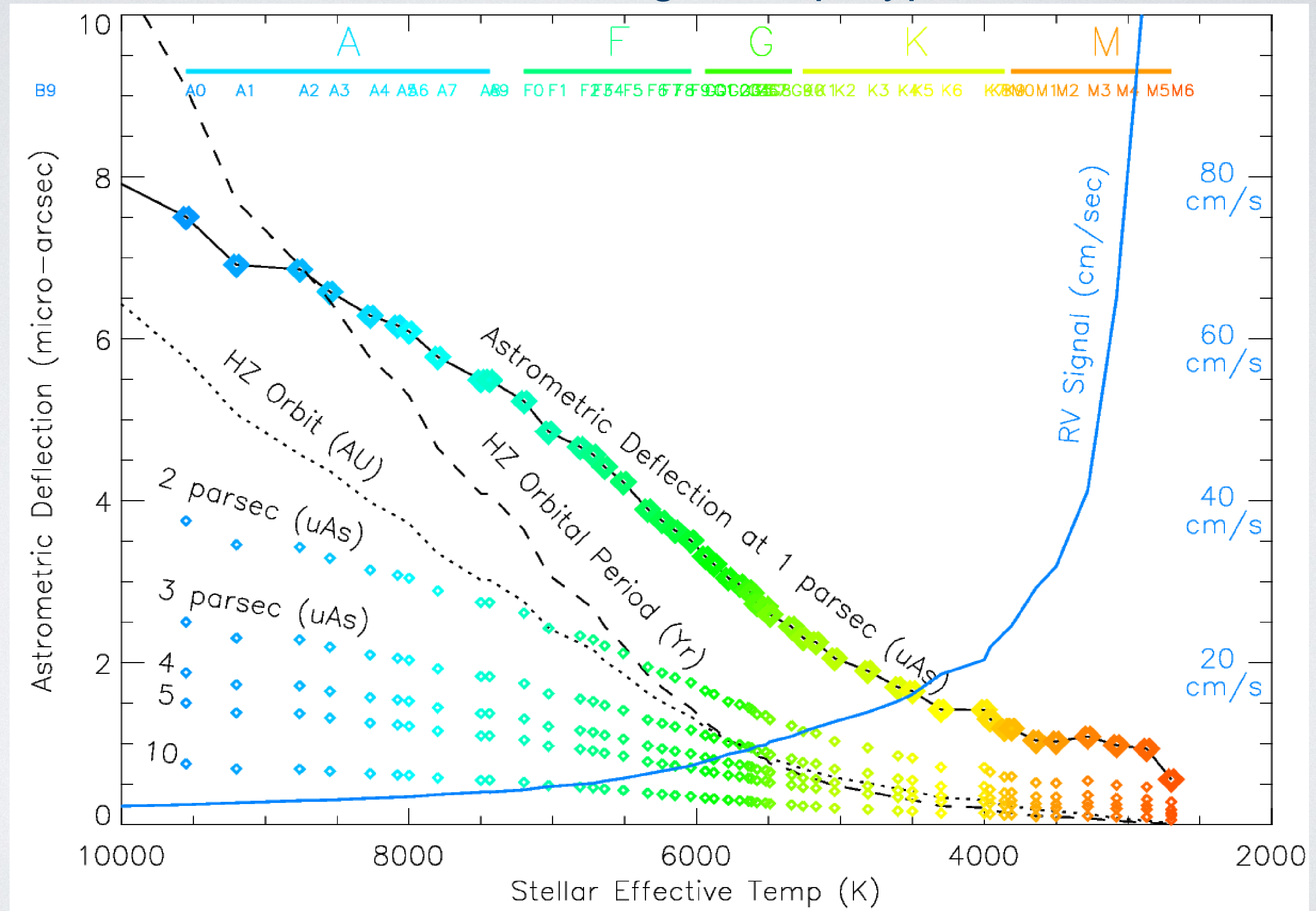
What is the stellar (and projected HZ rocky planet) population within 10 PC?

- 5 A-type Stars. likely 0 (or 1) HZ rocky planet
- 69 FGK-type Stars. Likely about a dozen HZ rocky planets
- 273 (+) M-type stars. Likely 140 HZ rocky planets

Locus of the Sun's motion due to planets (mainly Jupiter)



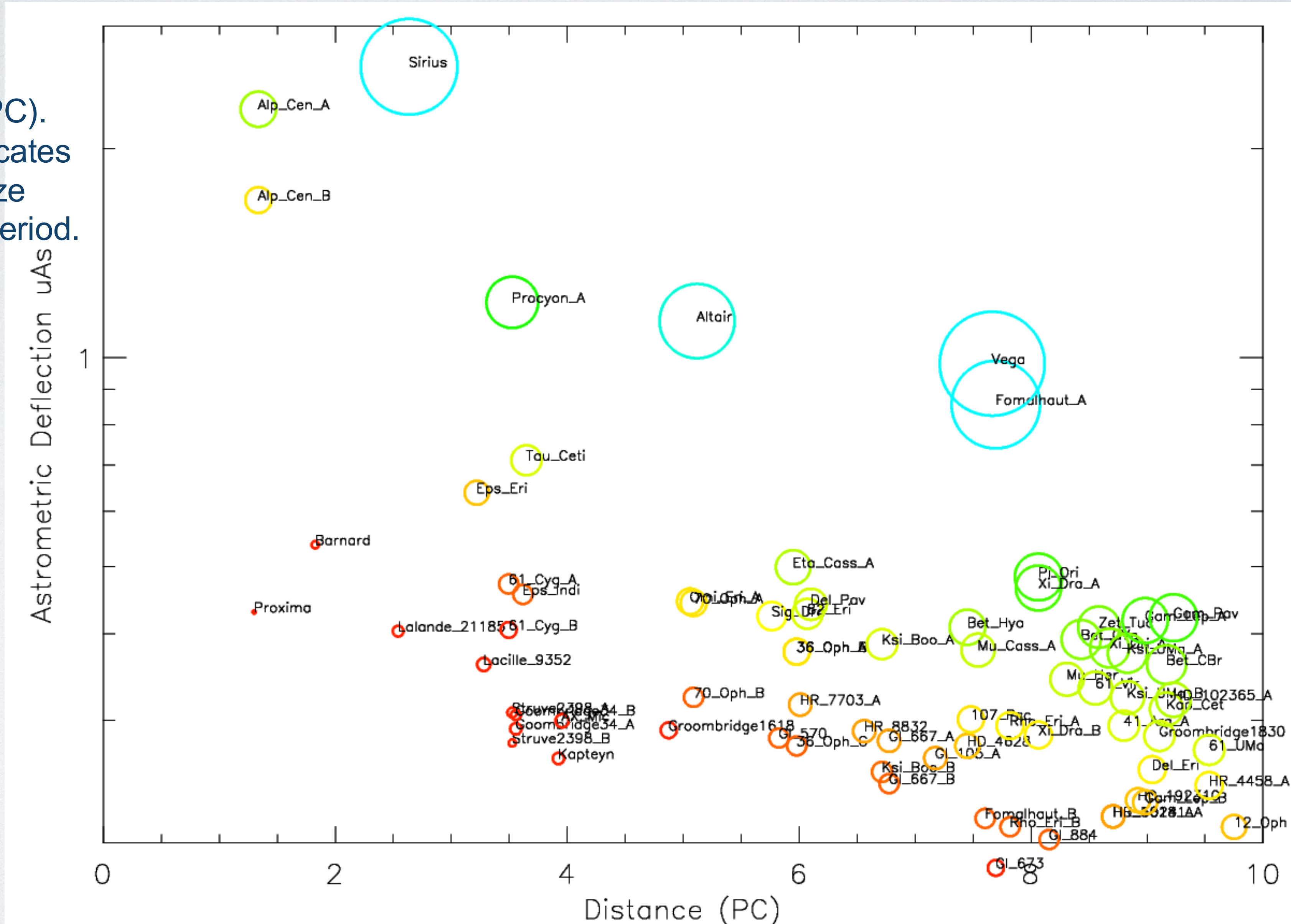
Astrometric signal of a 1 M_{earth} planet in the HZ assuming 1PC distant from star of given Sp. type.





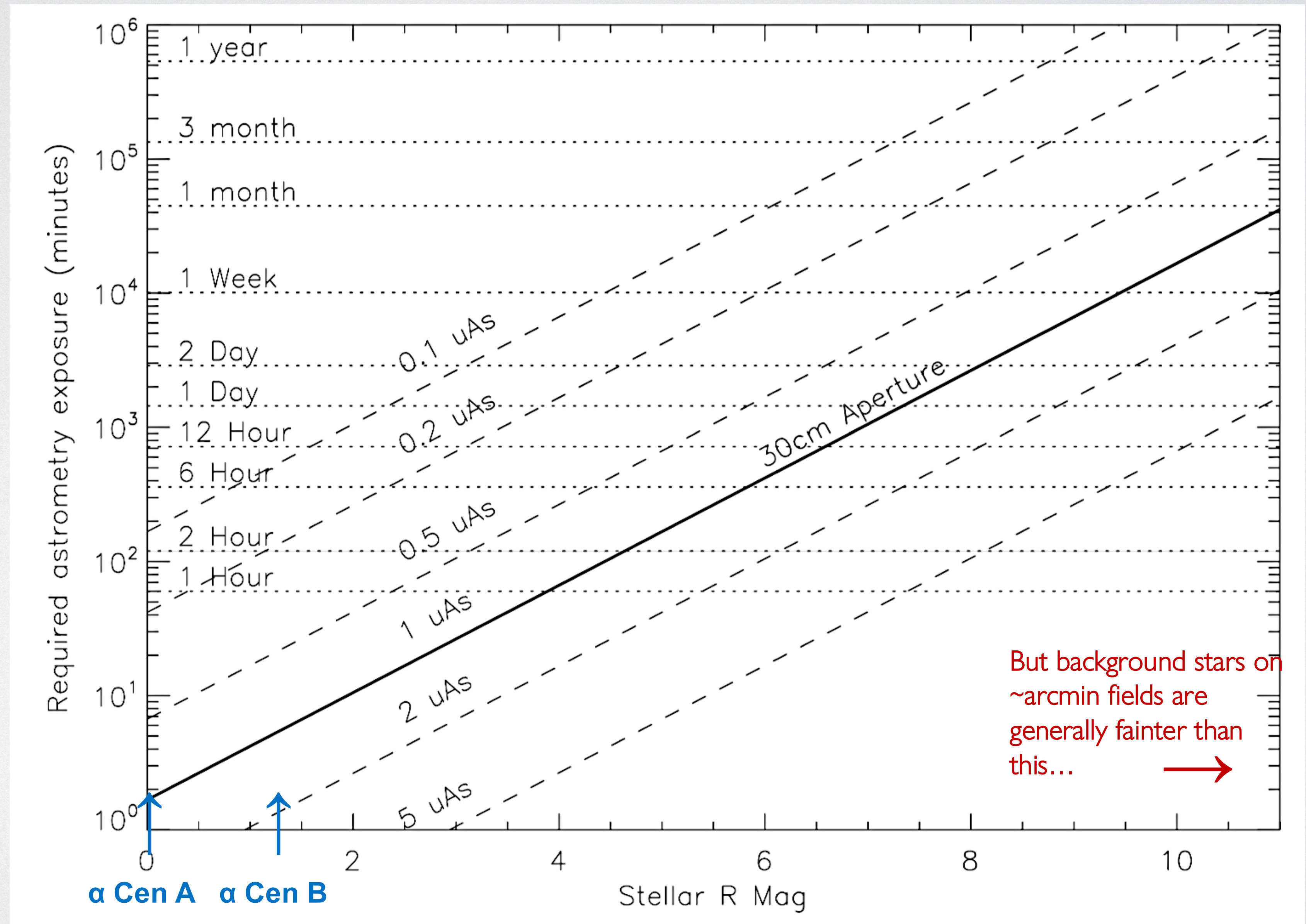
Astrometric signal for HZ Earth mass for all FGK stars to 10PC

Also M stars (<5PC).
Symbol color indicates host spectrum, size indicates orbital period.





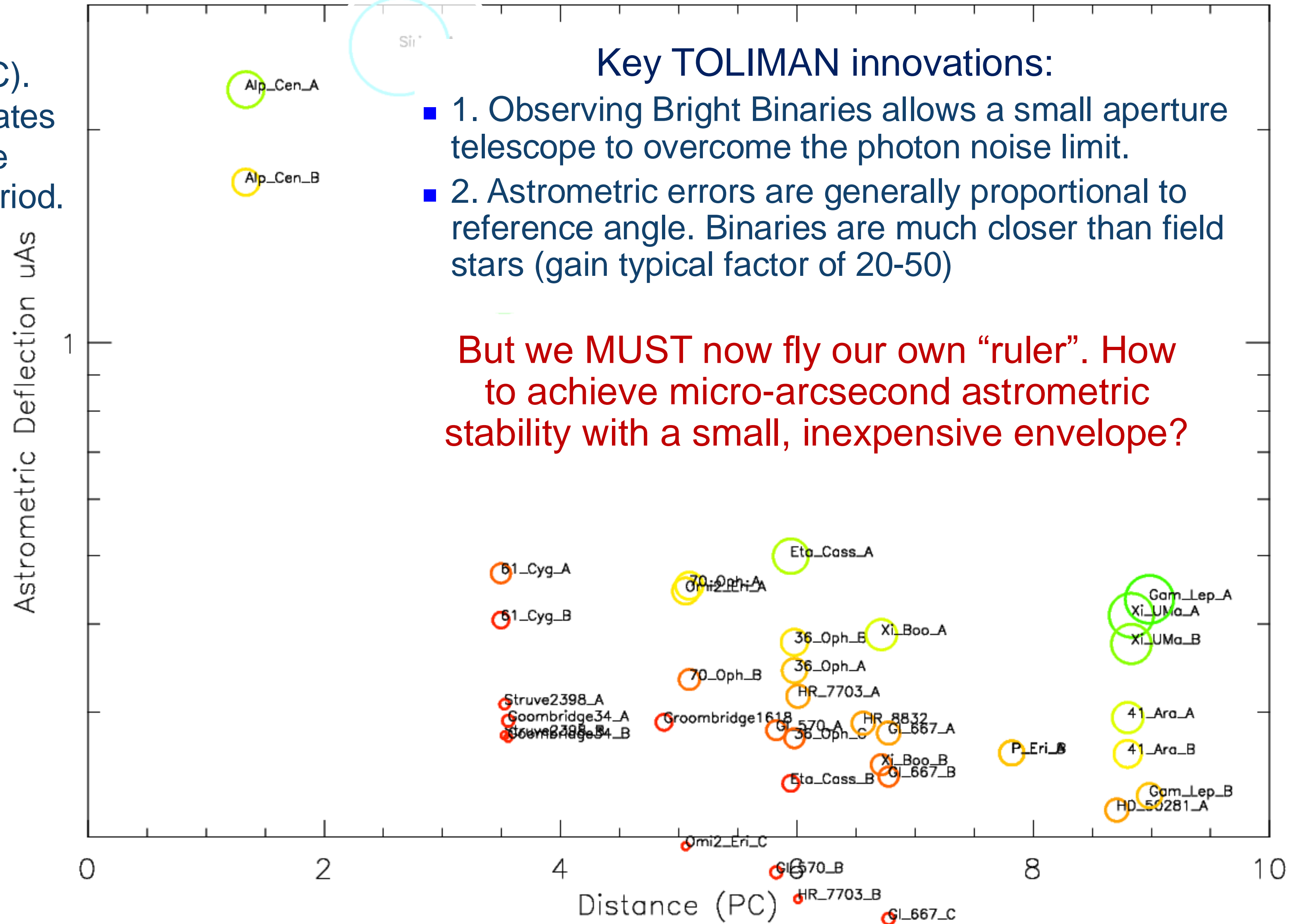
30cm Telescope fundamental (photon noise) limit: integration time required to obtain a given astrometric noise





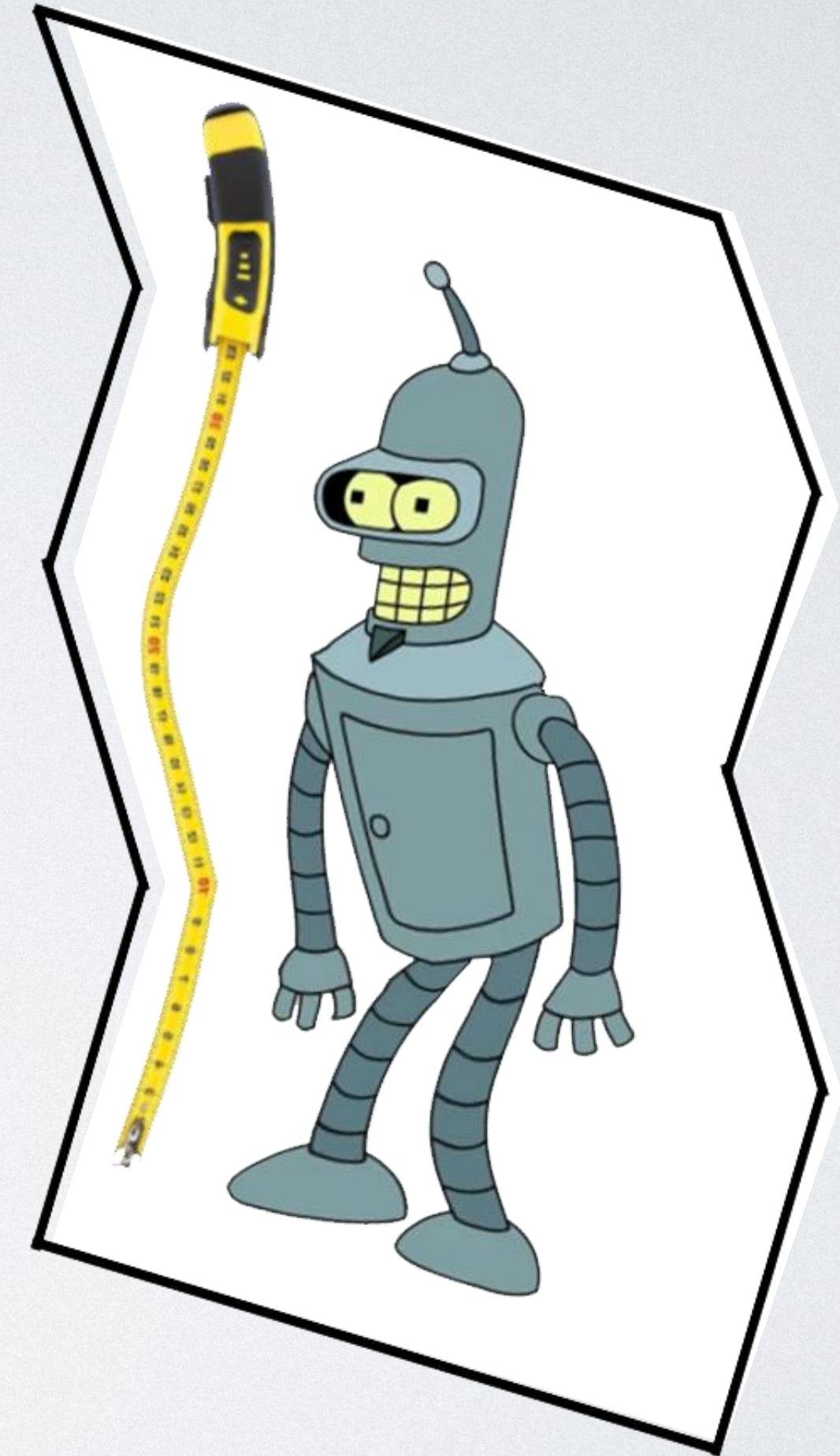
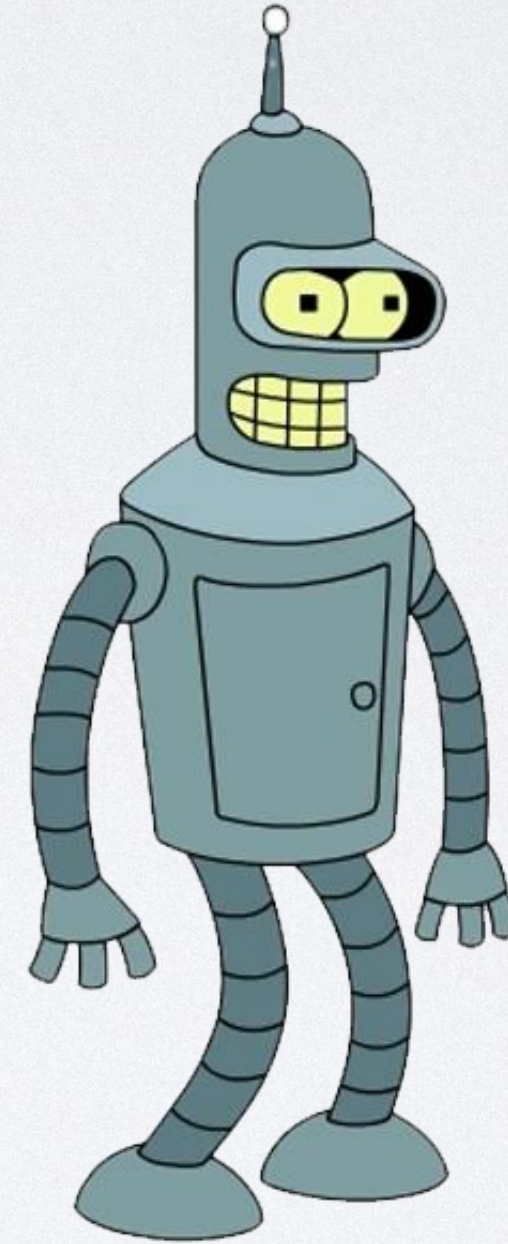
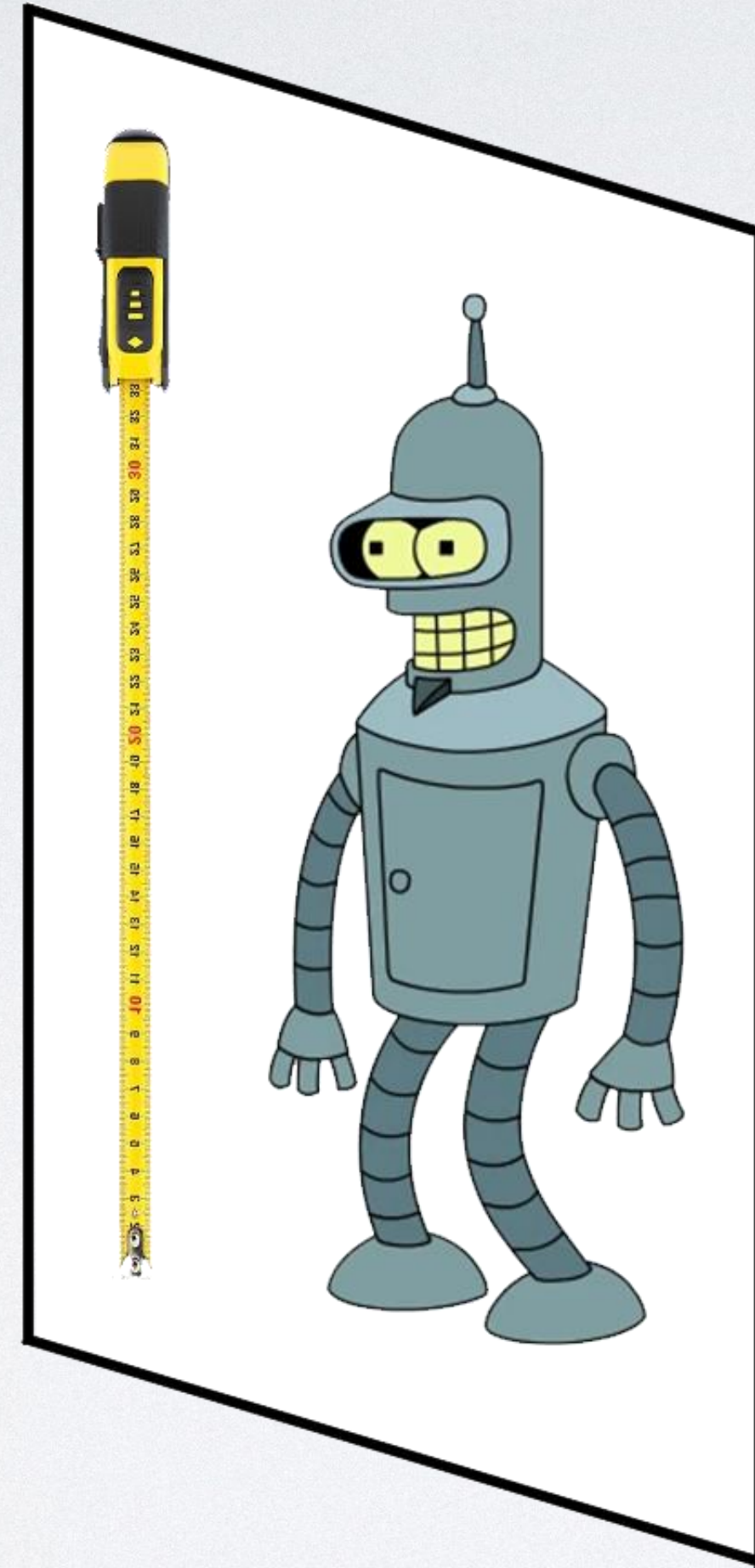
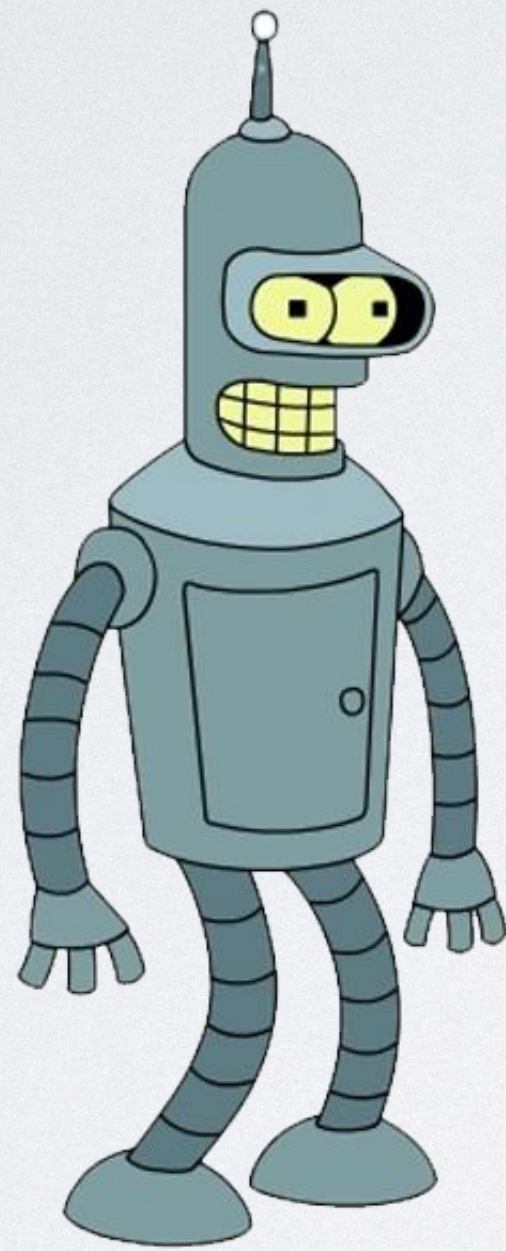
Astrometric signal for HZ Earth mass for all FGK stars to 10PC Binaries ONLY!

Also M stars (<5PC).
Symbol color indicates host spectrum, size indicates orbital period.





Optical distortions don't matter if they bend both your ruler and your object...

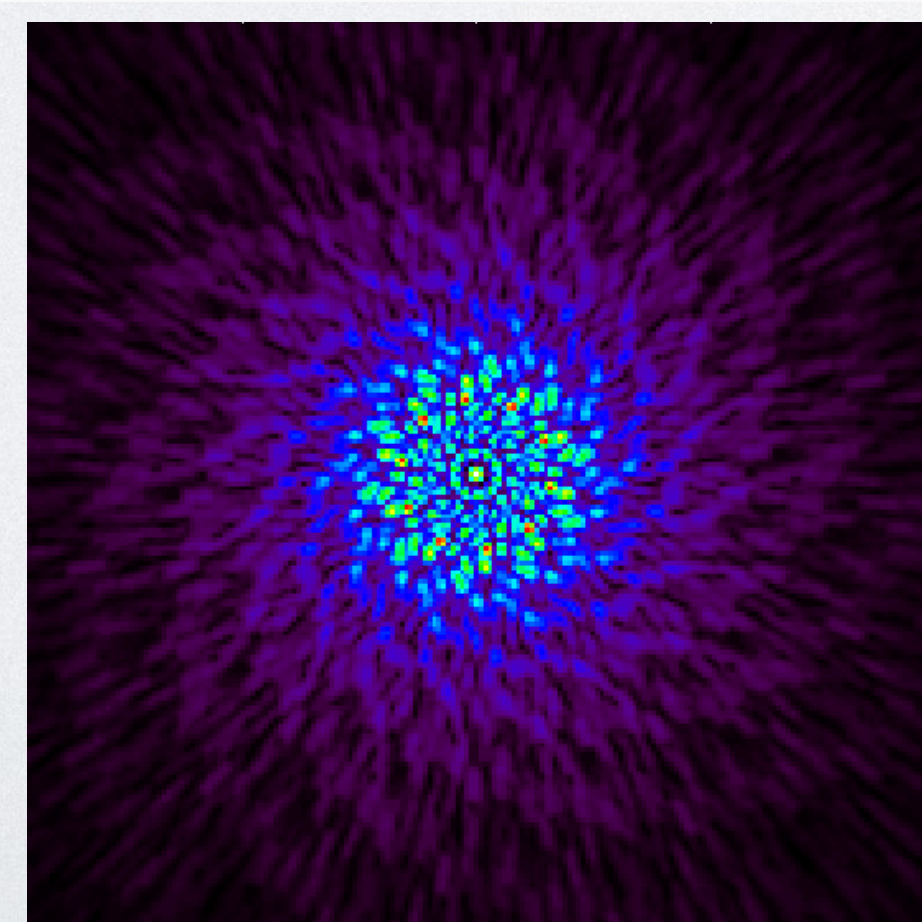
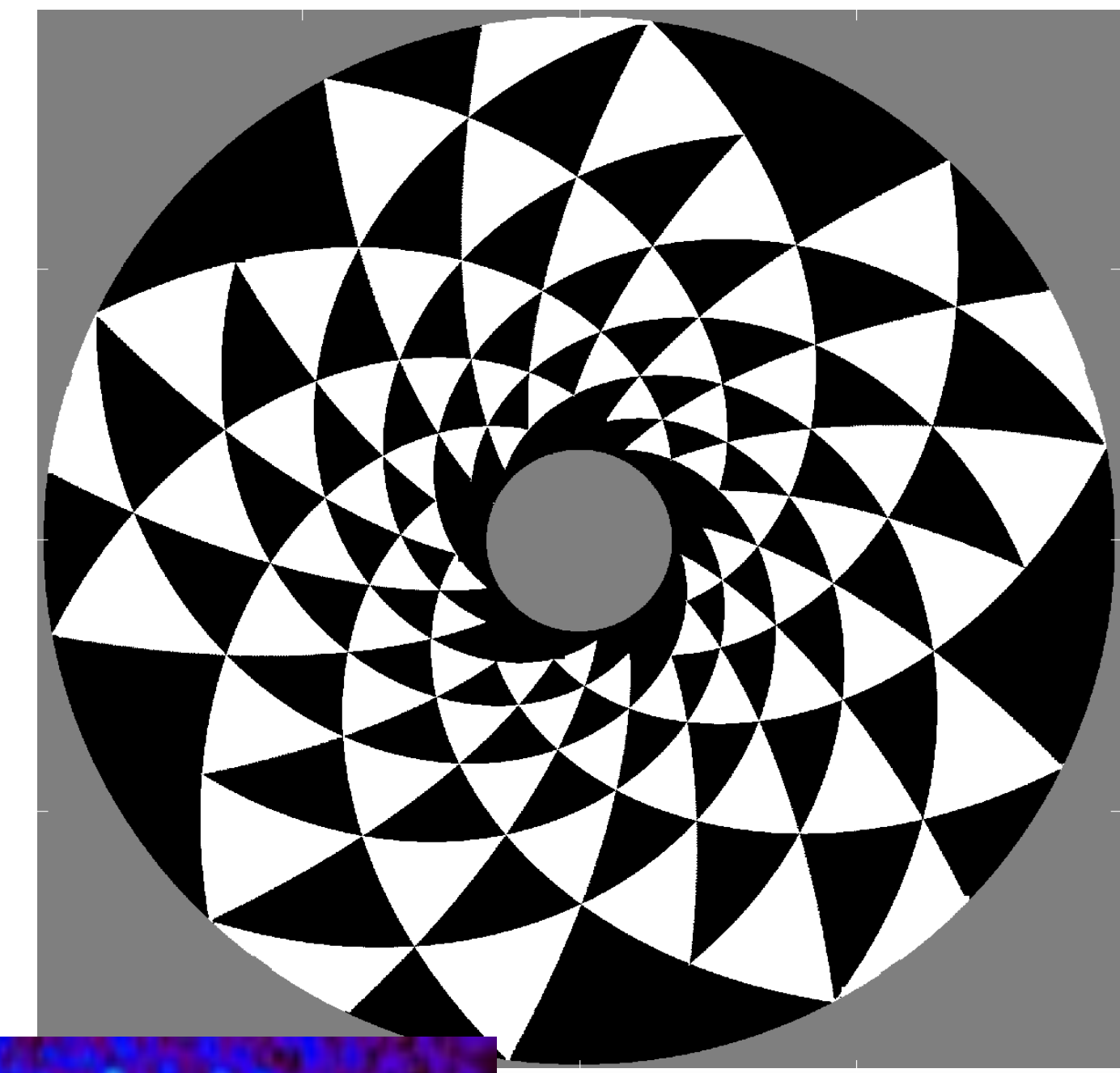
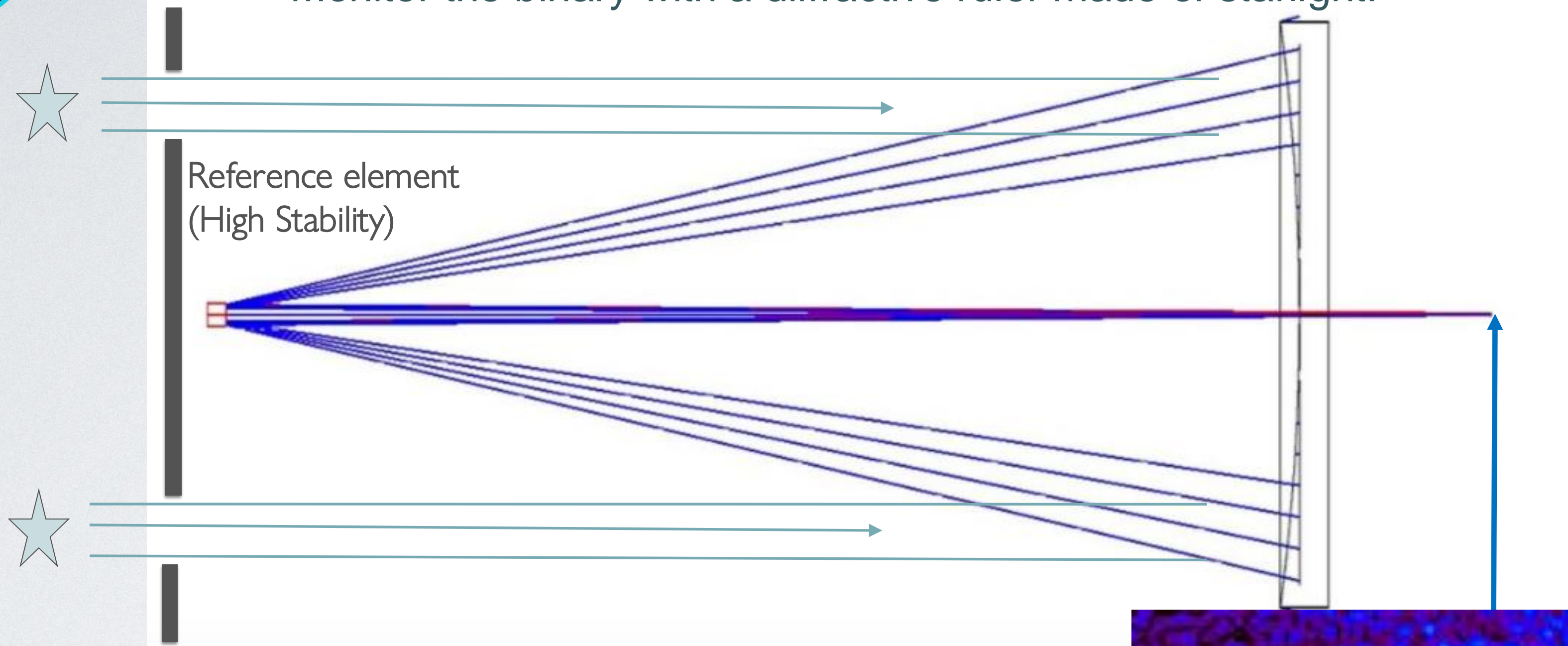




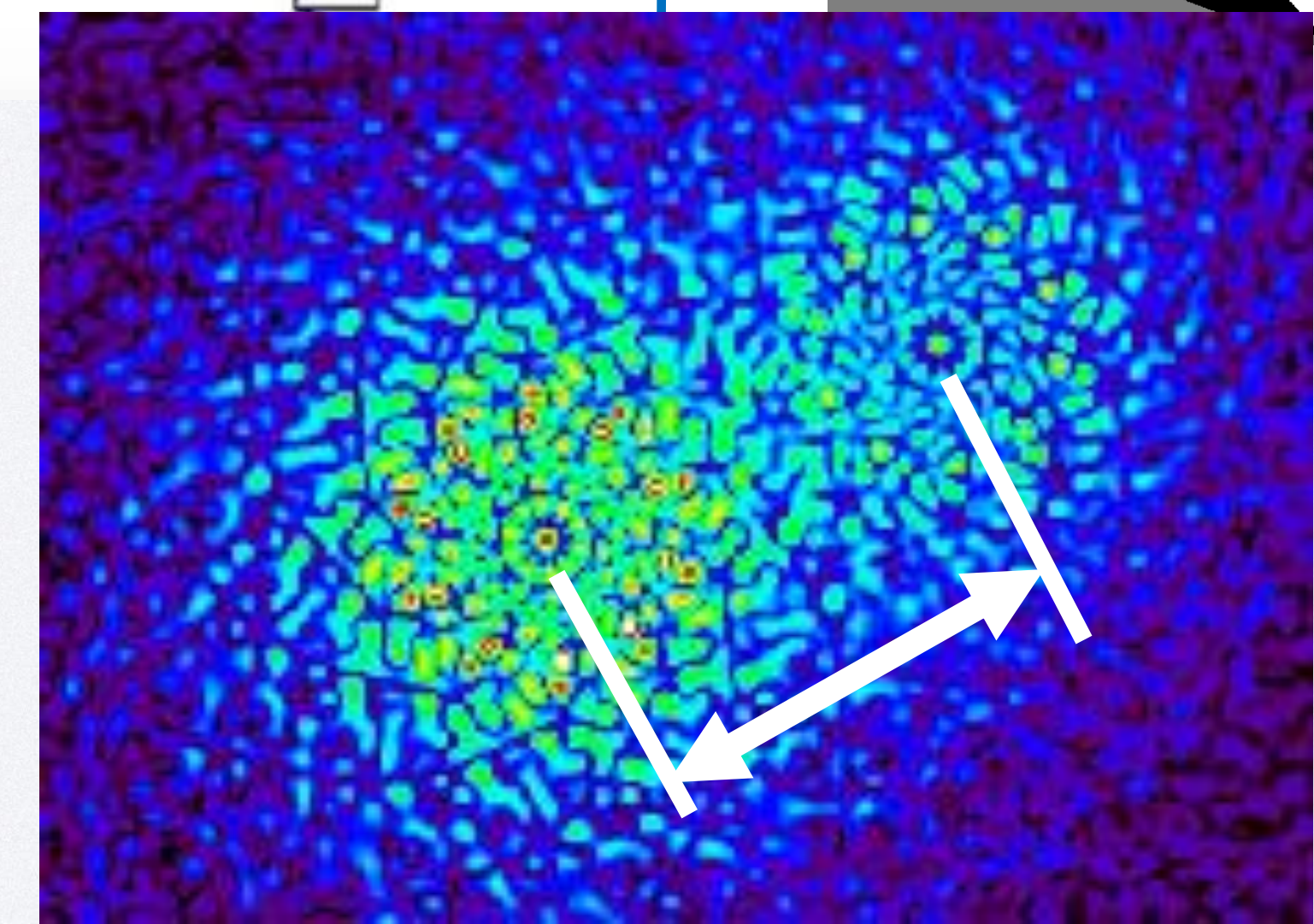
Optical concept empowering TOLIMAN

Diffractive-pupil pupil mask

Monitor the binary with a diffractive ruler made of starlight!

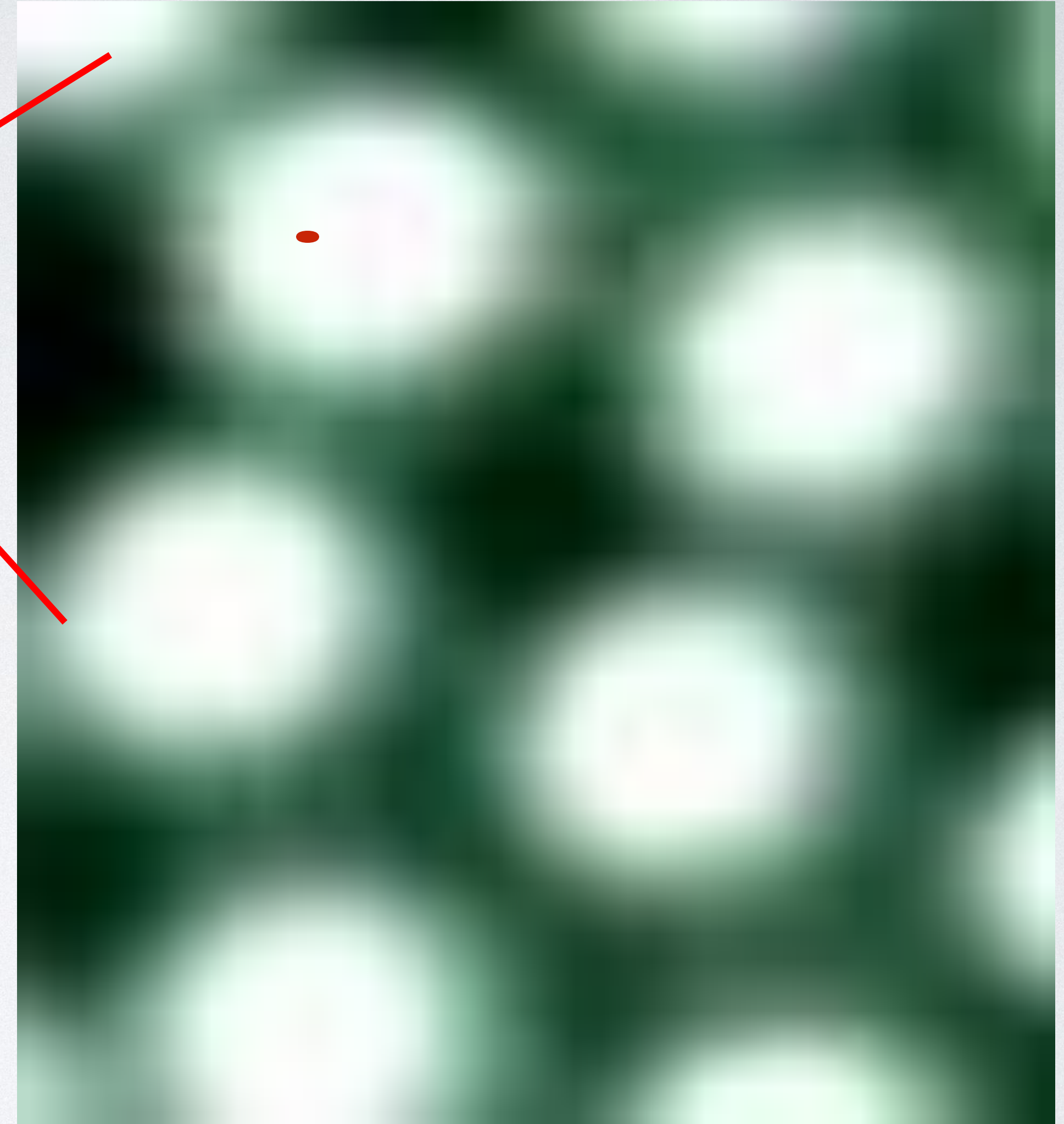
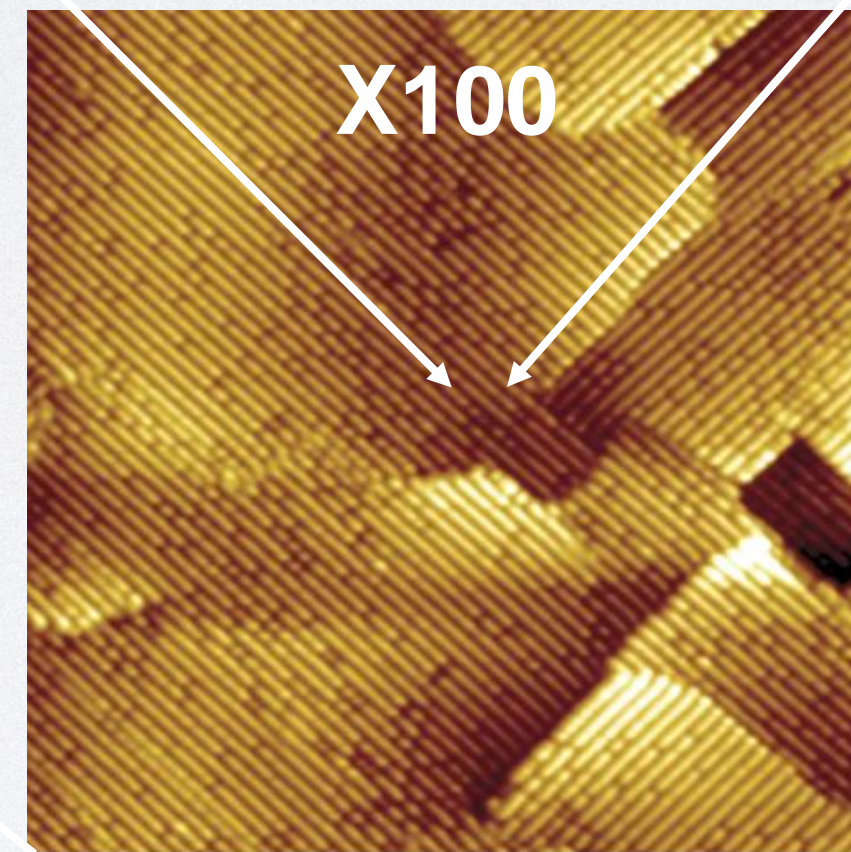
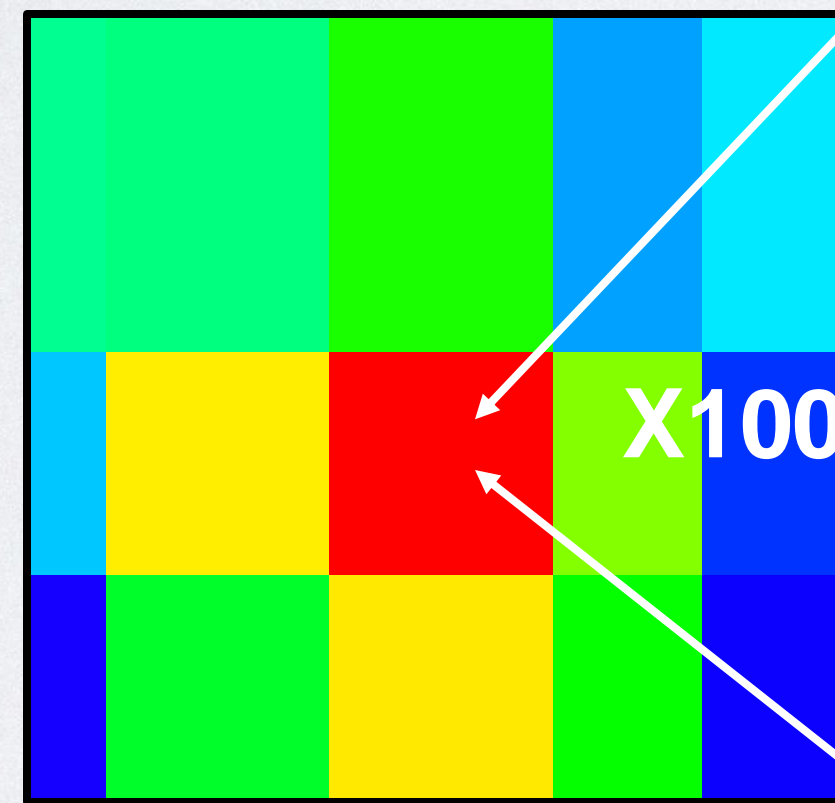
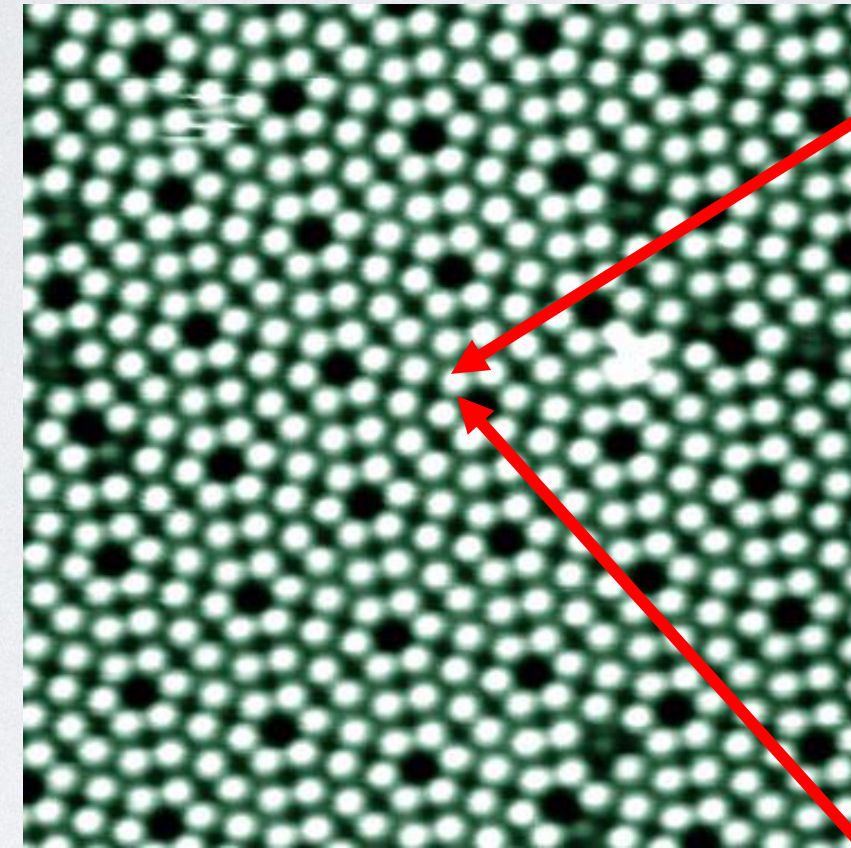
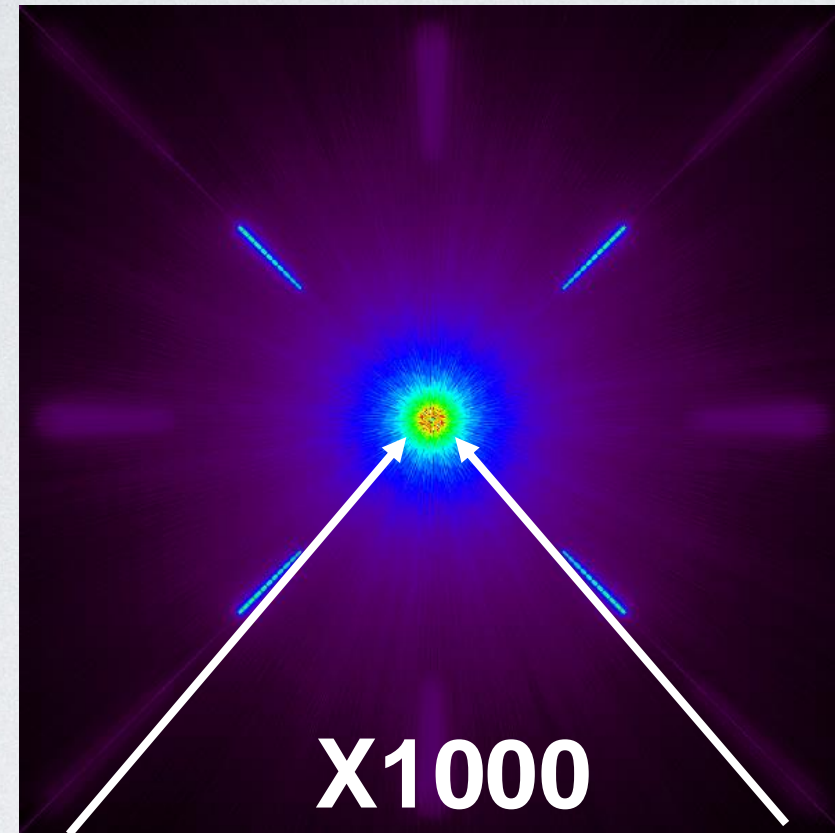


Point
Spread
Function





The scale of the Instrumental Challenge





Key TOLIMAN innovations:

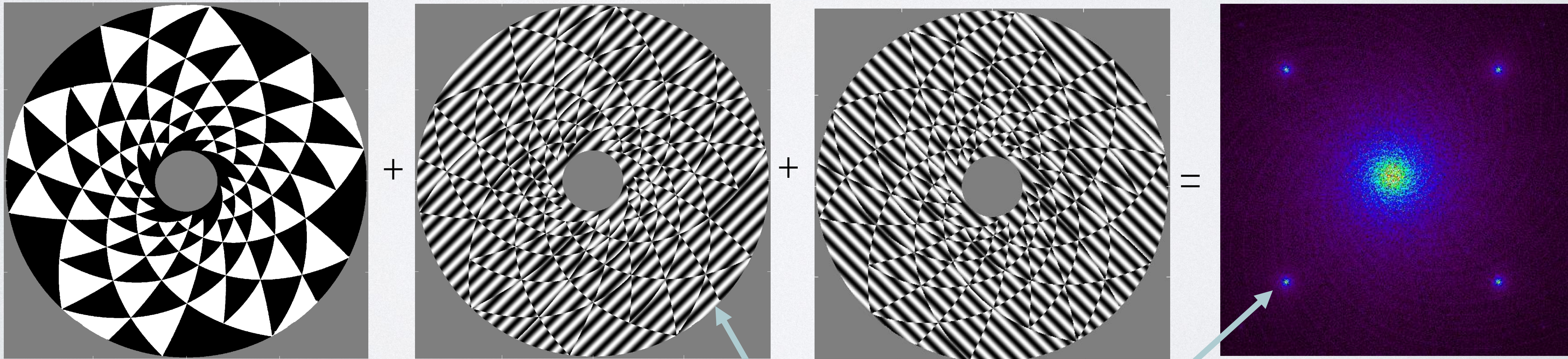
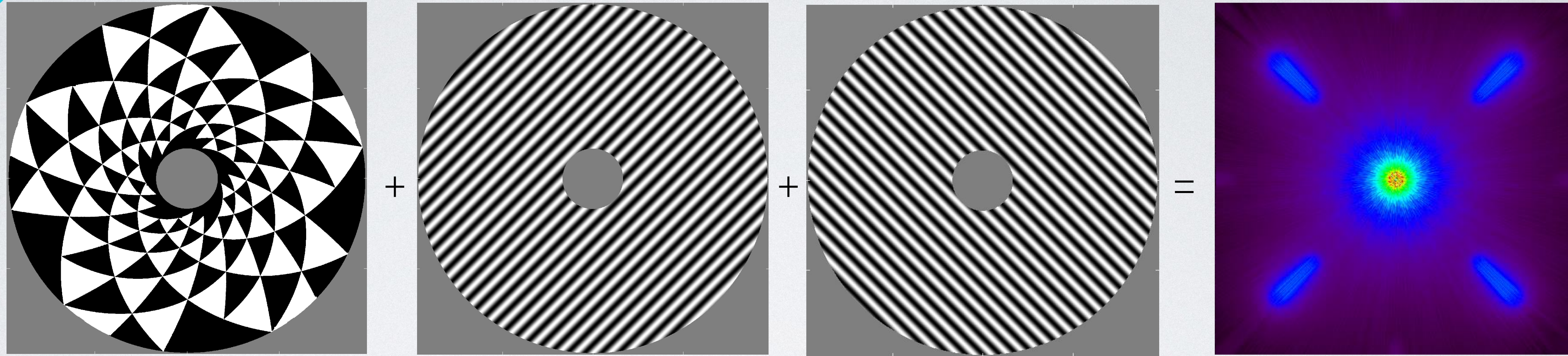
- 1. Observing Bright Binaries allows a small aperture telescope to overcome the photon noise limit.
- 2. Astrometric errors are generally proportional to reference angle. Binaries are much closer than field stars (gain typical factor of 20-50)
- 3. The Diffractive Pupil removes most error terms arising from distortion in the optical train. The fundamental ruler element can be made monolithic, thermally stable, and precisely monitored.
- 4. Naturally spreads the starlight over many pixels, preventing detector saturation and at the same time giving major statistical benefits in beating down noise

But there is a flaw!

Our ruler is made of light (fringes) depends on the effective wavelength of the starlight – which varies with star T_{eff} !



Nailing down the wavelength: adding a spectrometer (... and Jedi Fourier mind tricks)

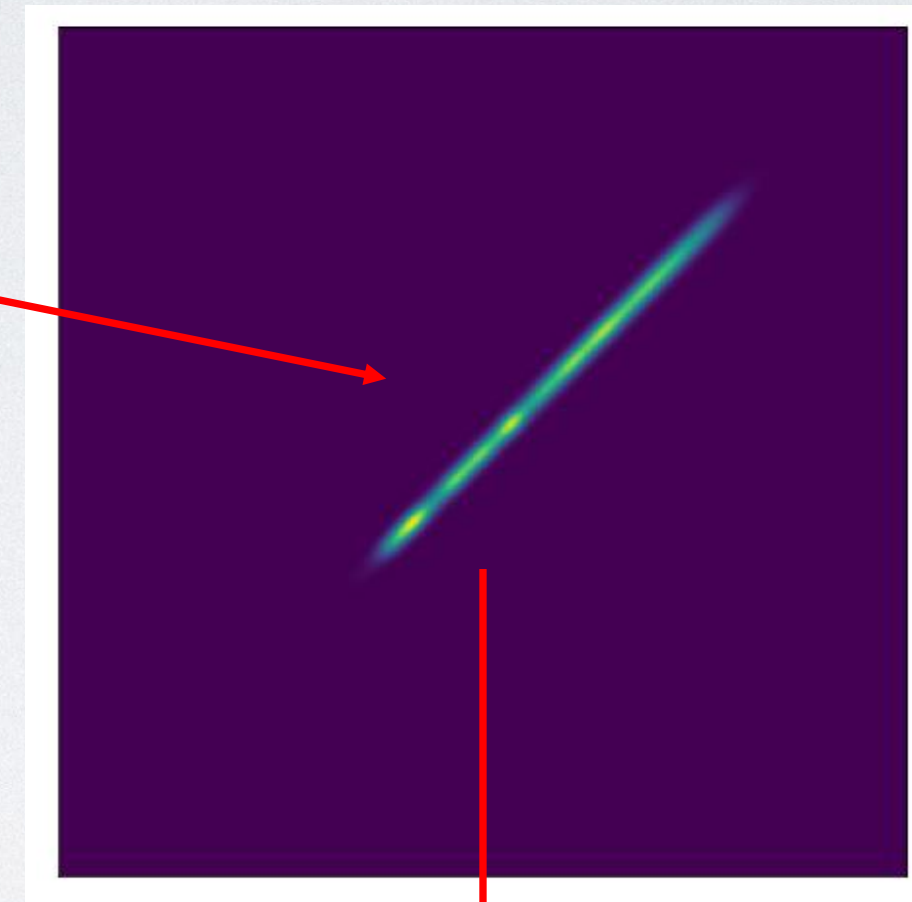
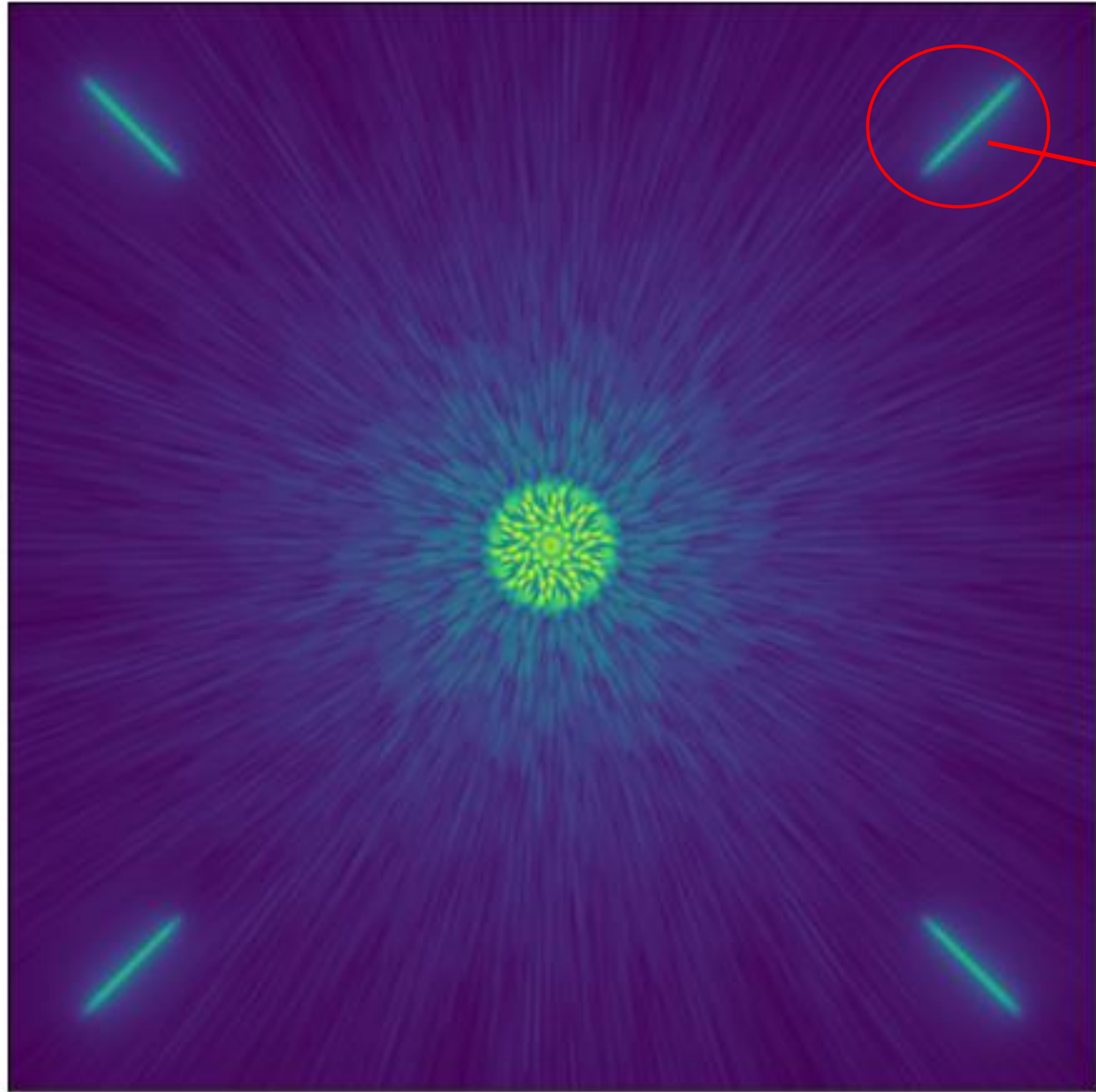


Inverted phase grating

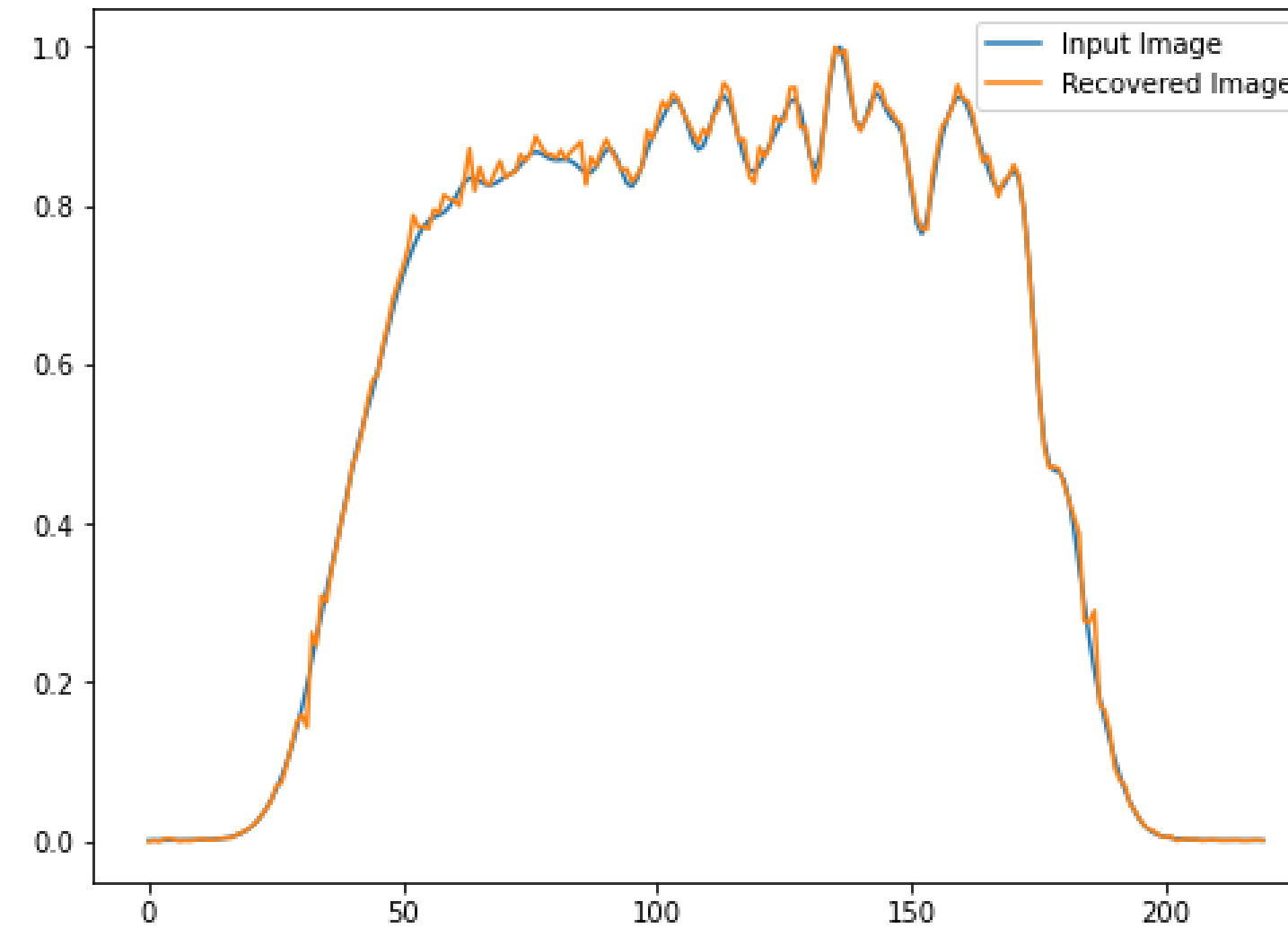
Ideal Spectrometer "slit" function



IN-BUILT SPECTROMETER



Diagonals





HOW DO WE BUILD ALL THIS?

- Implement the novel optical principles in hardware
- Excellent pointing (low jitter – better than 1” per second image drift)
- Extreme mechanical and thermal stability required
- Challenges also for on-board data processing, downlink.
- Everything must fit within a cubesat form factor
- Hardest of all - fit within a very aggressive budget!



Liquid Crystal Phase Holograms



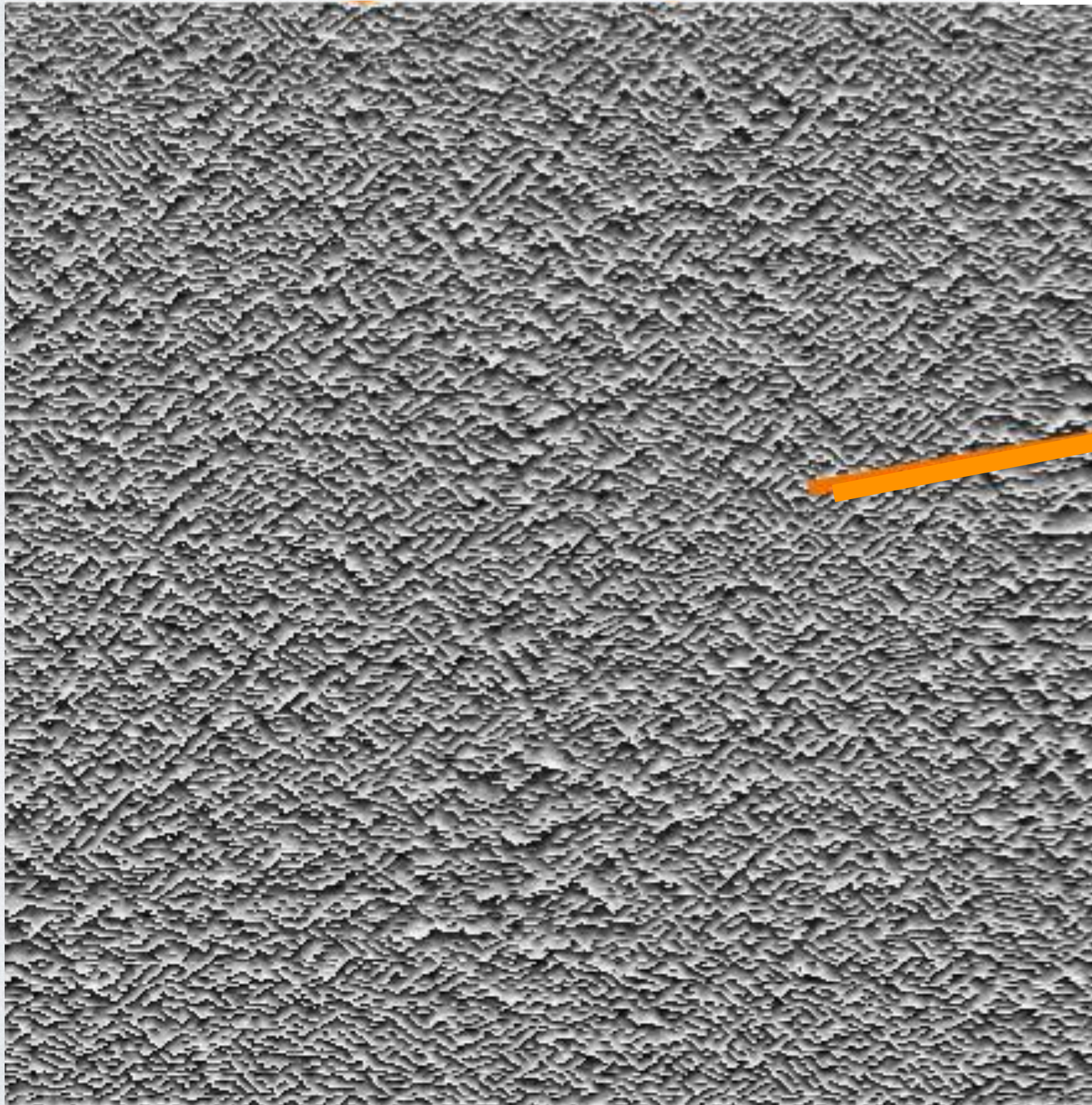
Frans Snik



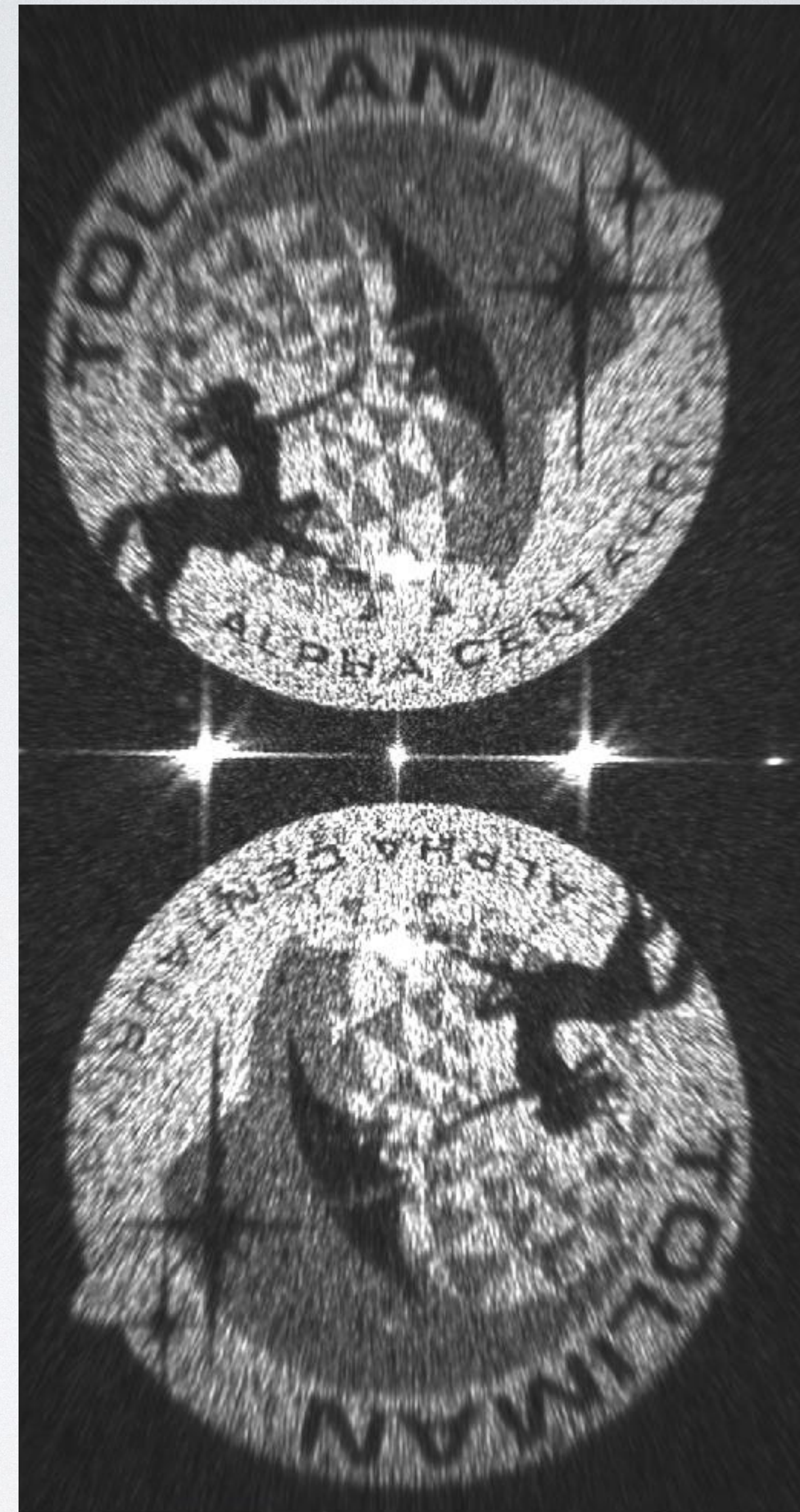
David Doelman



Universiteit
Leiden



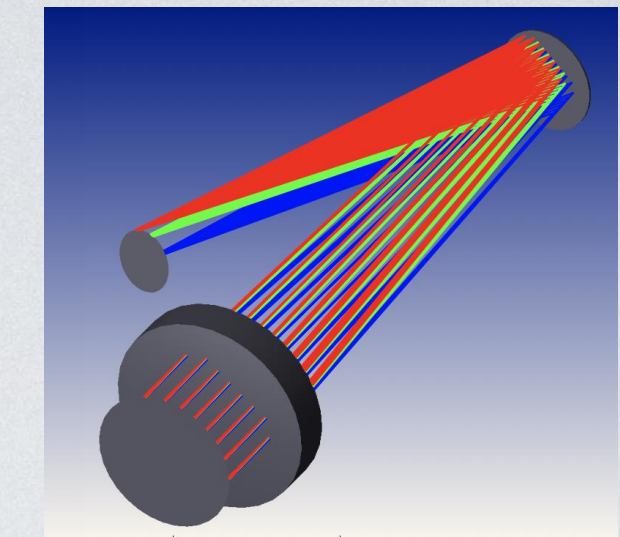
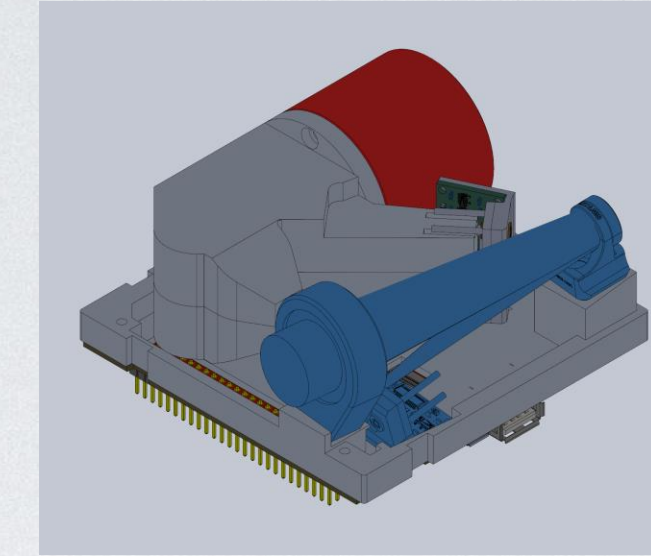
Measured PSF



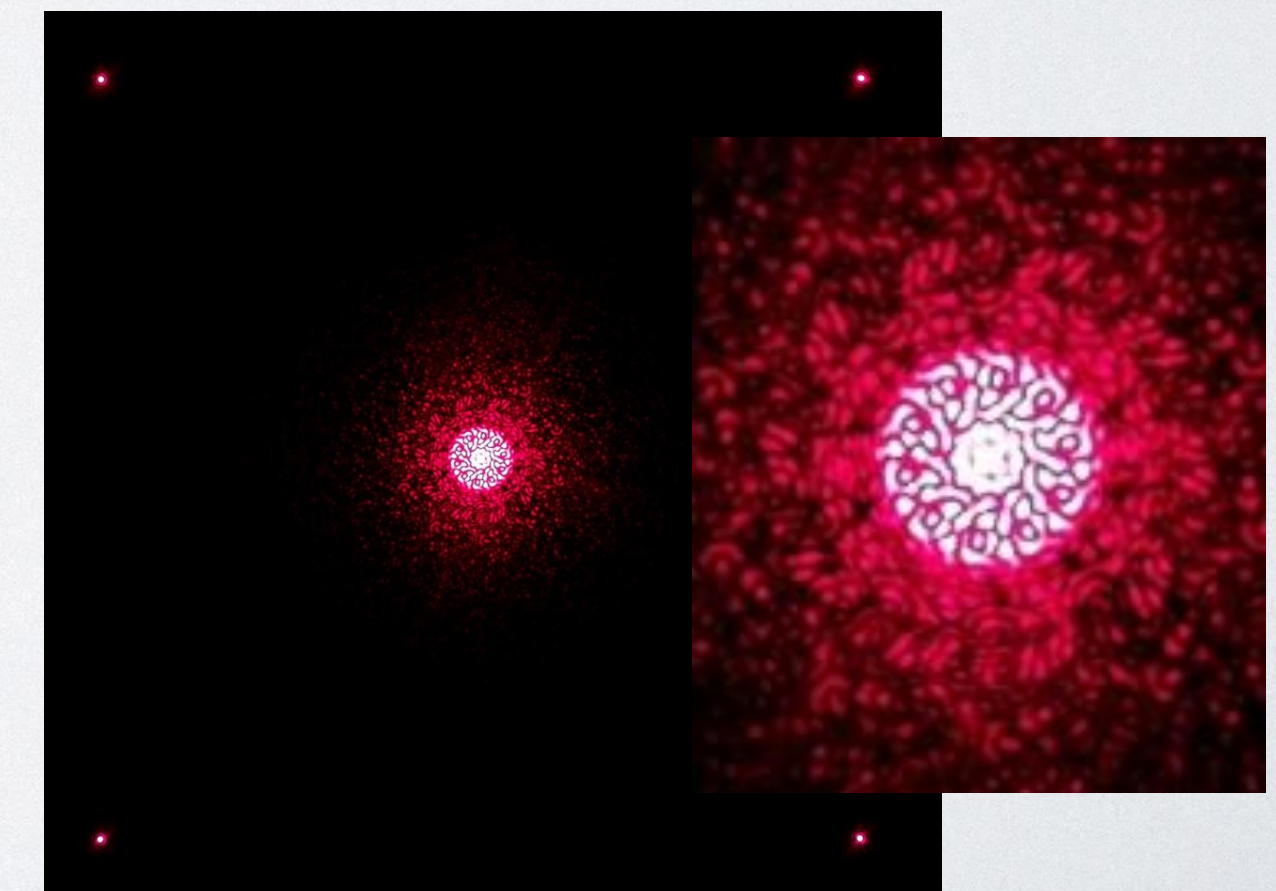
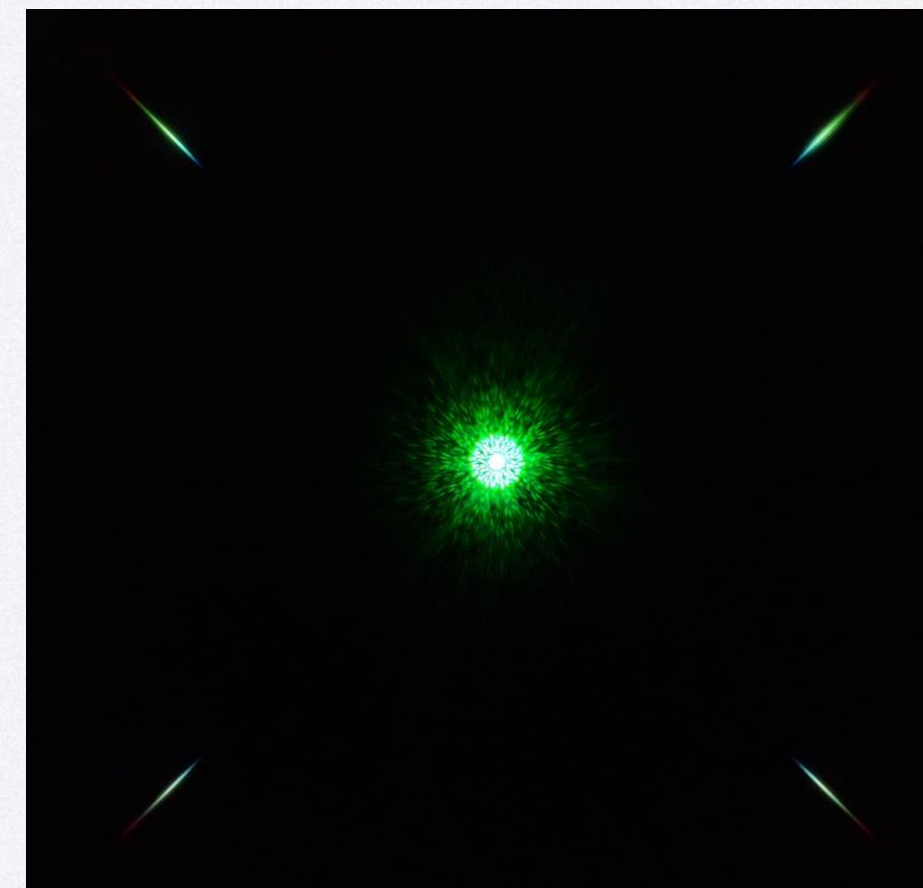
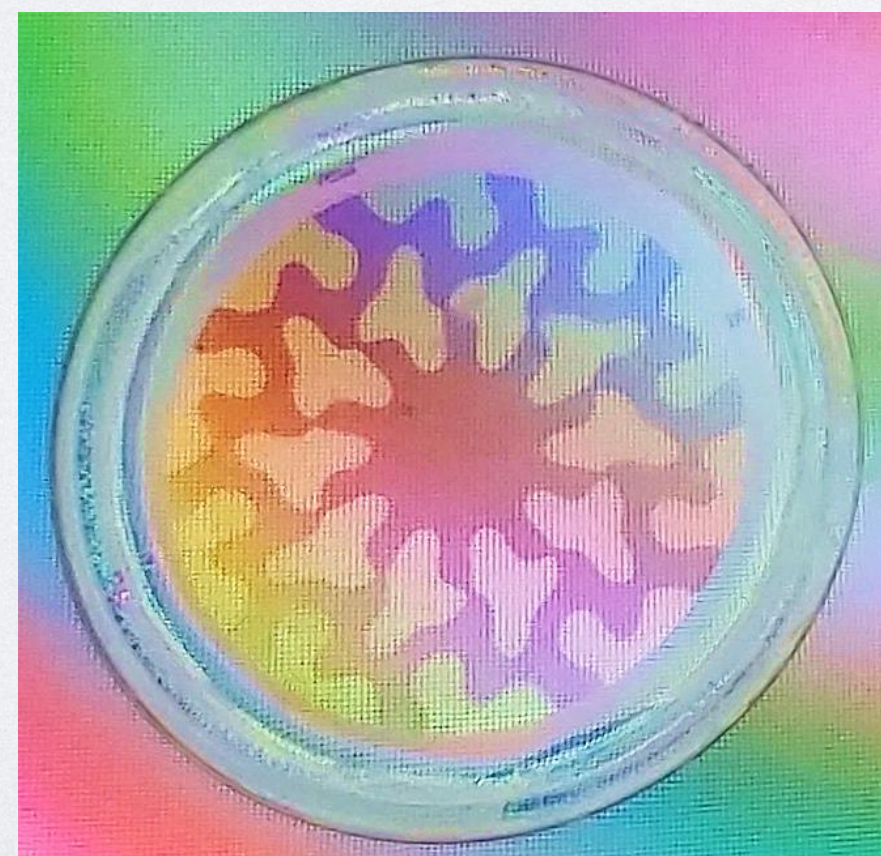
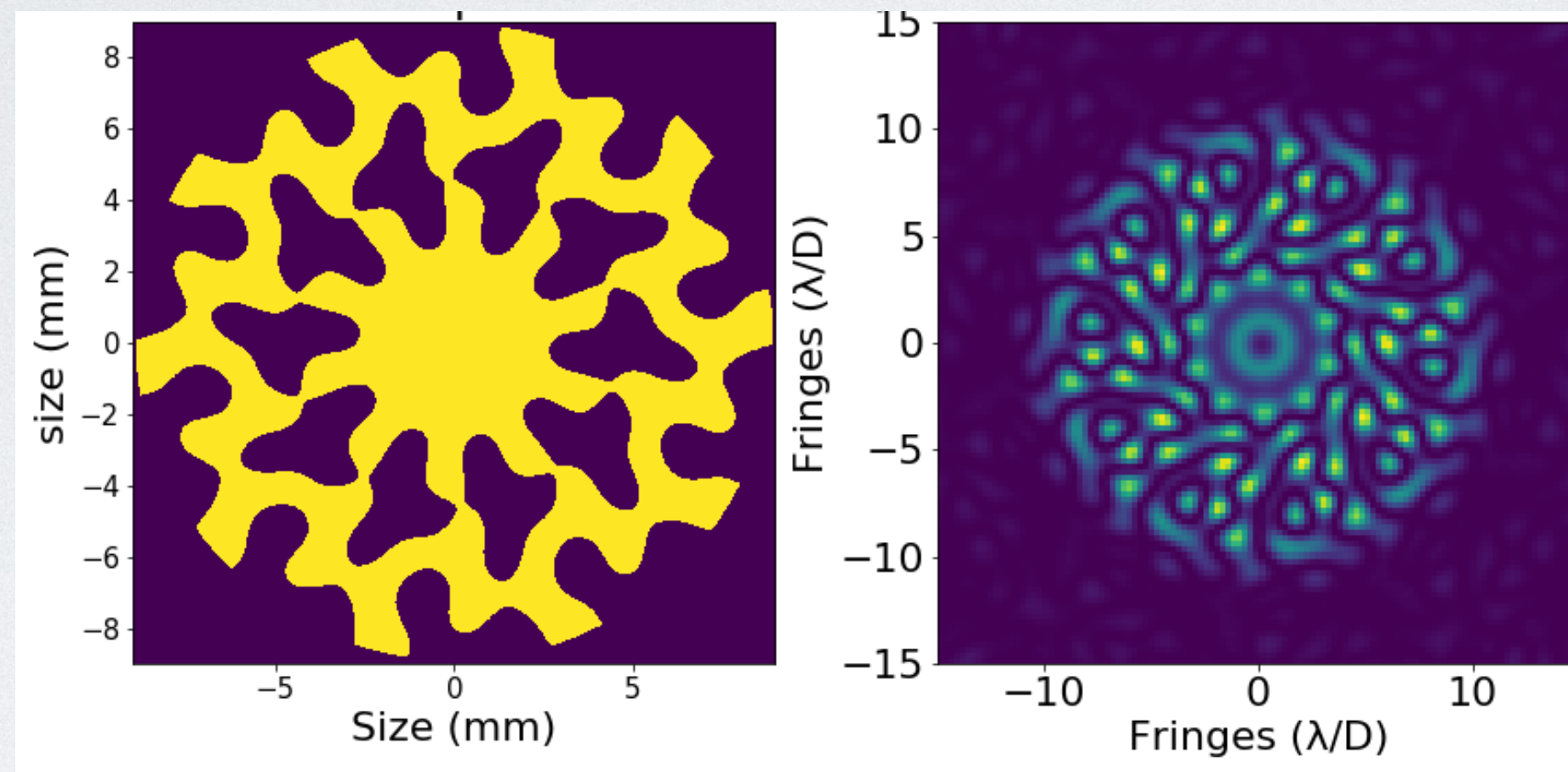


Program: TinyTol, Toliman, ...

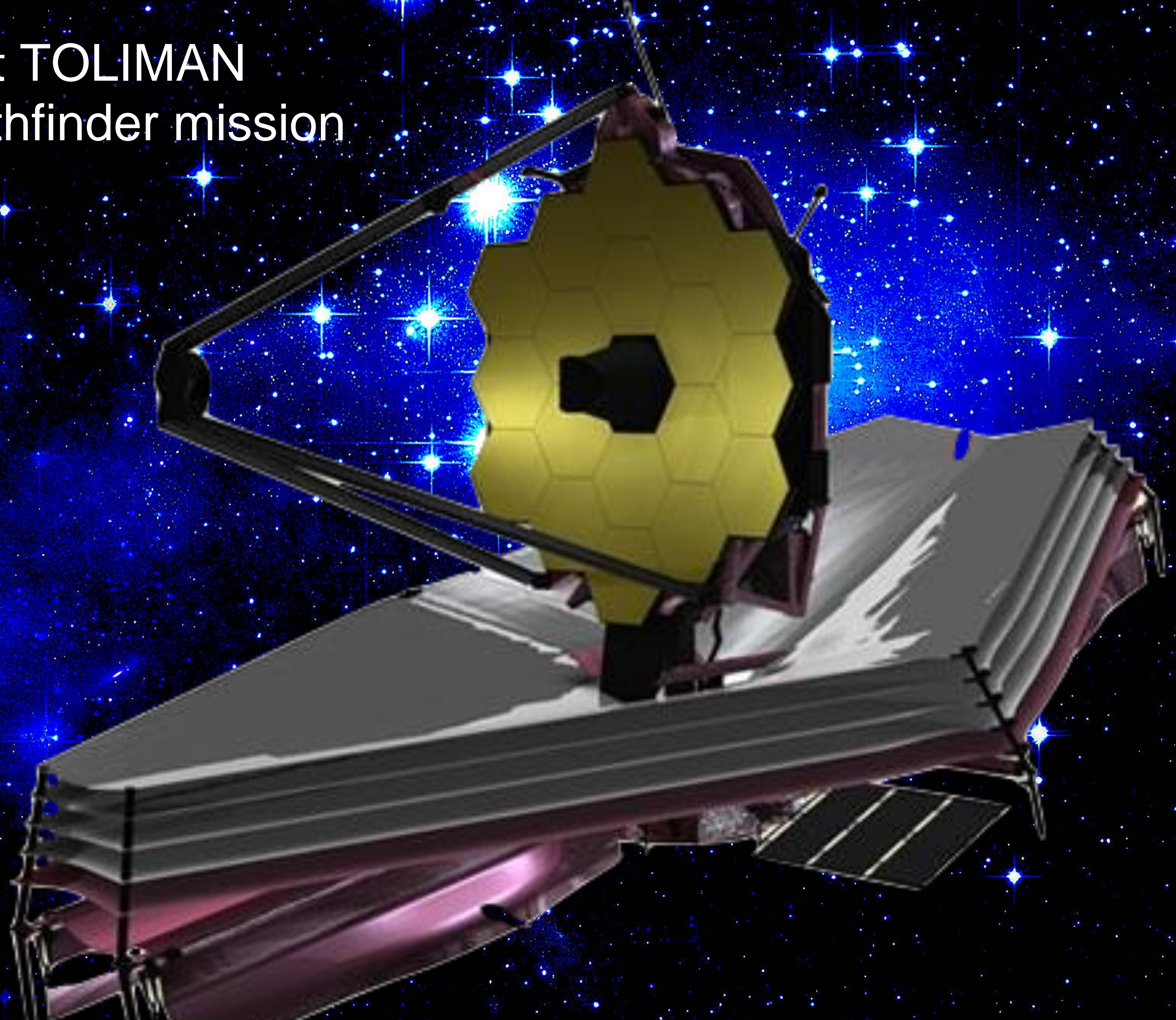
- TinyTol flew aboard CUAVA1
 - 3U cubesat. ISS resupply deployed 2020
 - 20mm aperture, $f=15\text{cm}$, 1 deg FoV
 - Consumer-grade electronics
 - Built, deployed, flown. No data.



TT instrument / Ray Trace

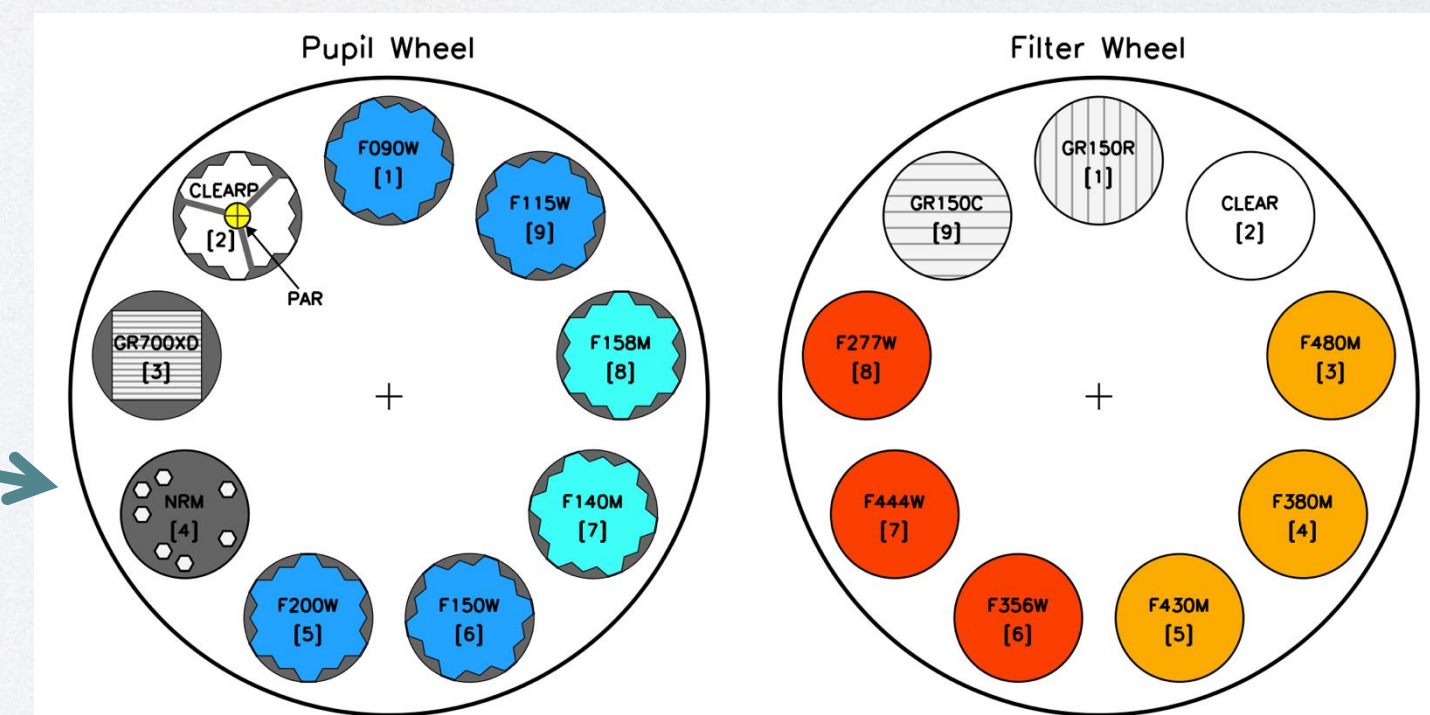
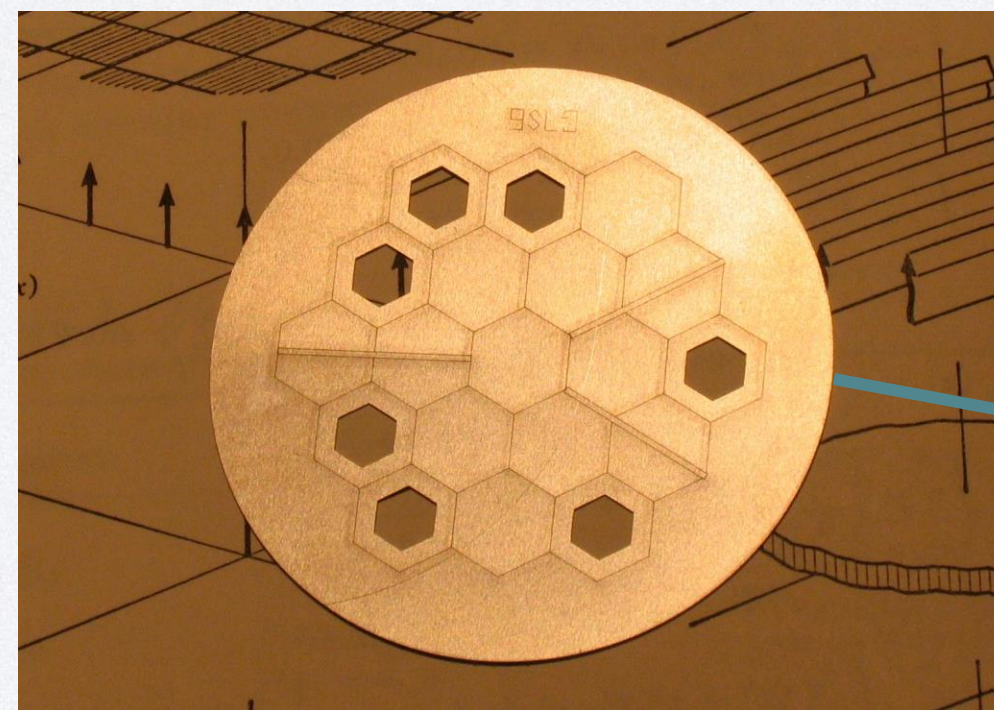
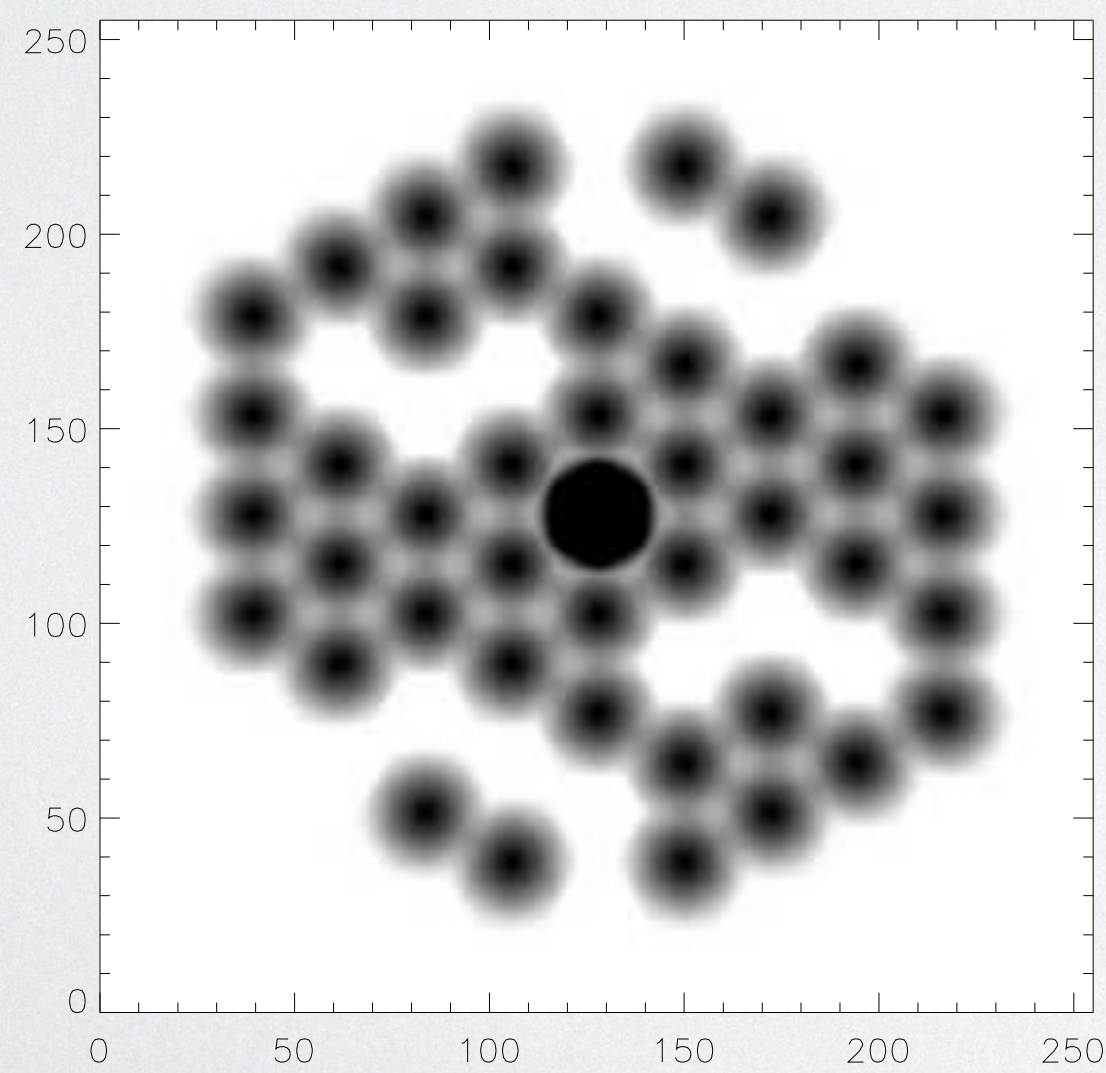
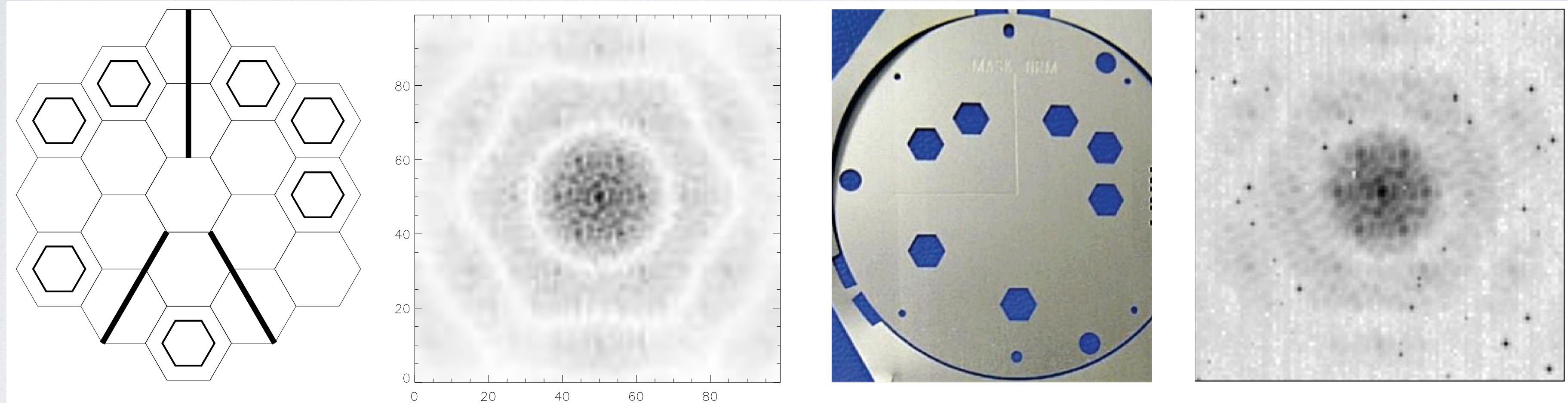


A Modest TOLIMAN Precursor/Pathfinder mission

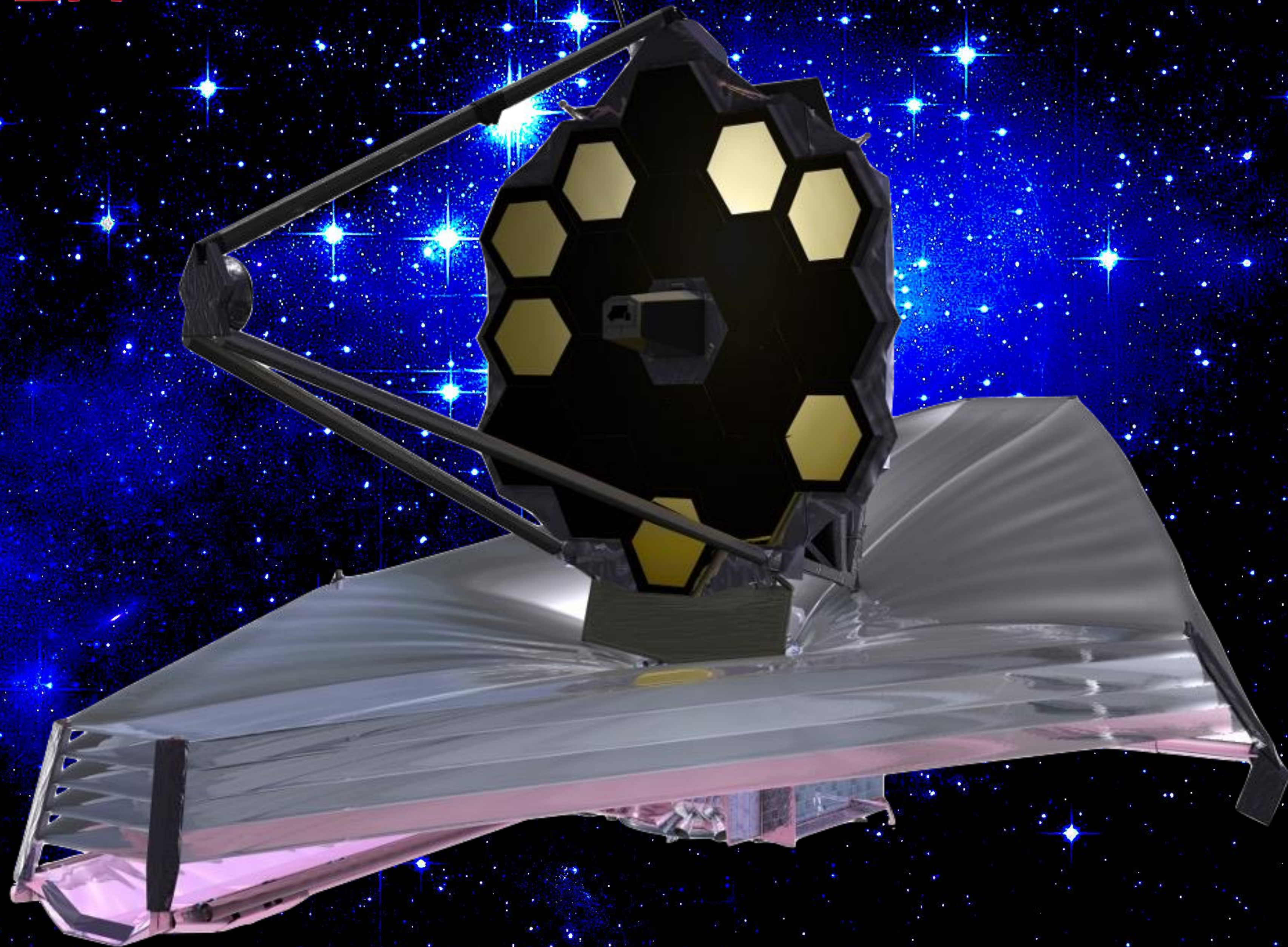


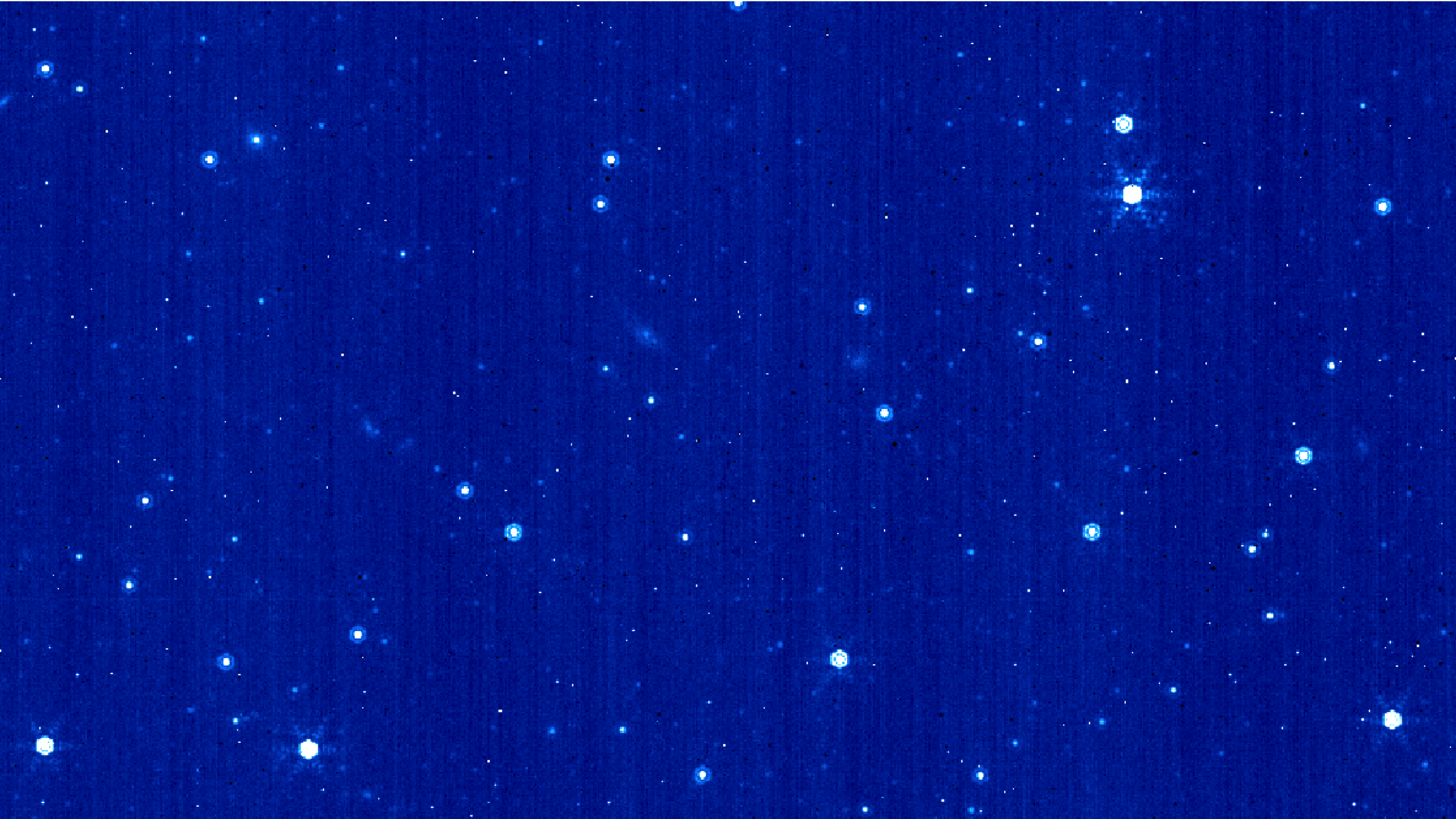


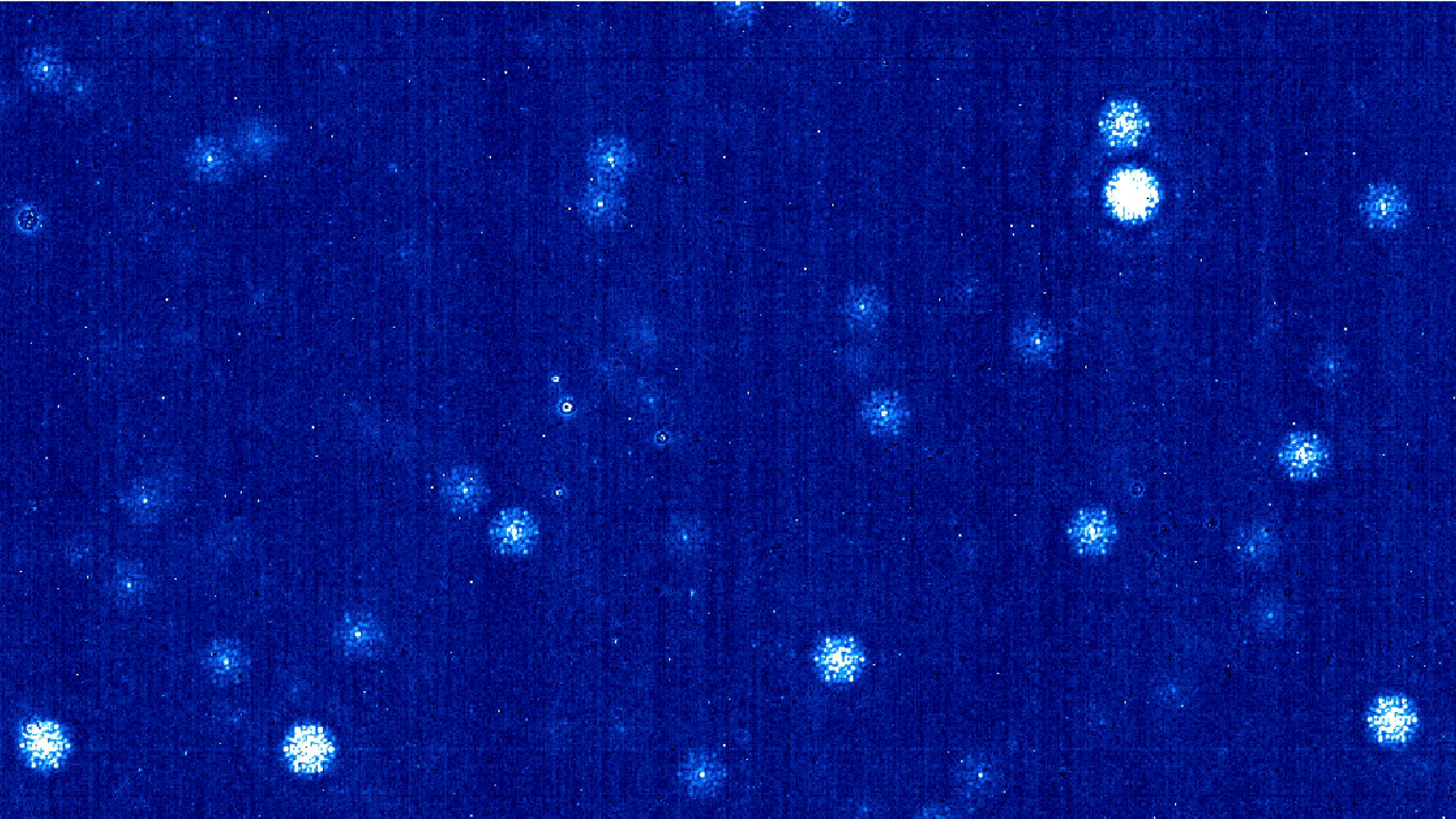
NIRISS AMI: THE JWST INTERFEROMETER



Thanks @ **NASA**





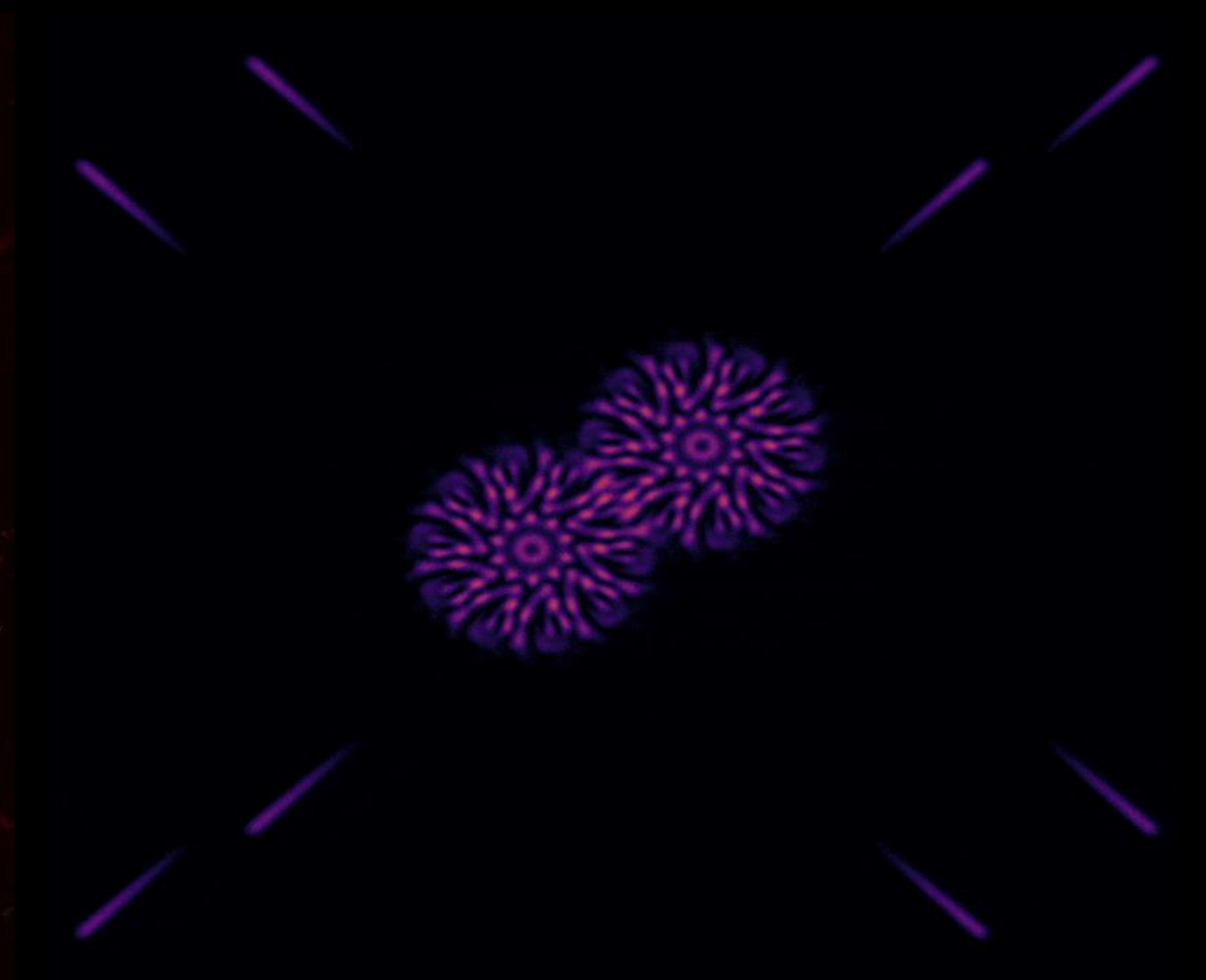
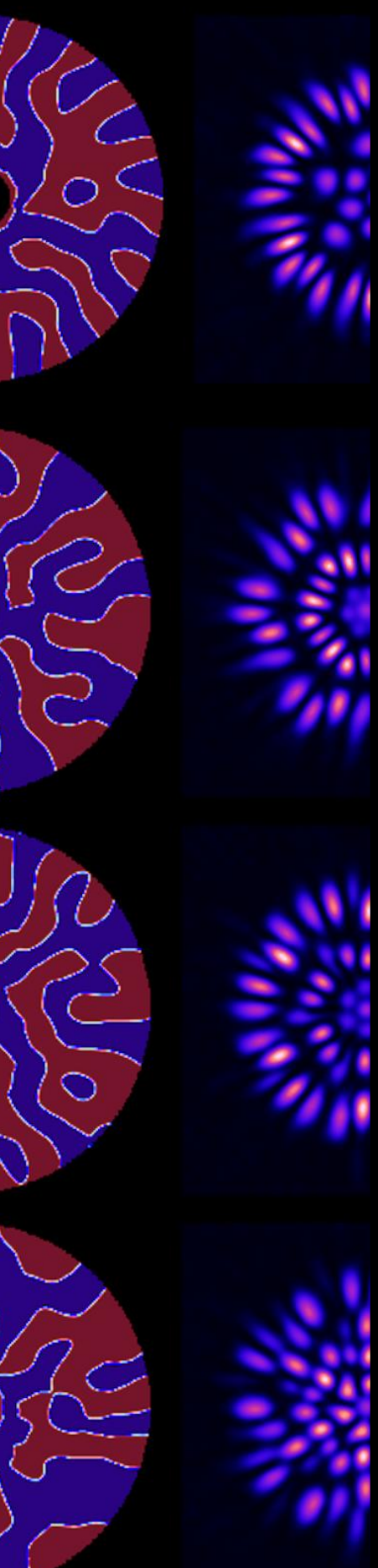




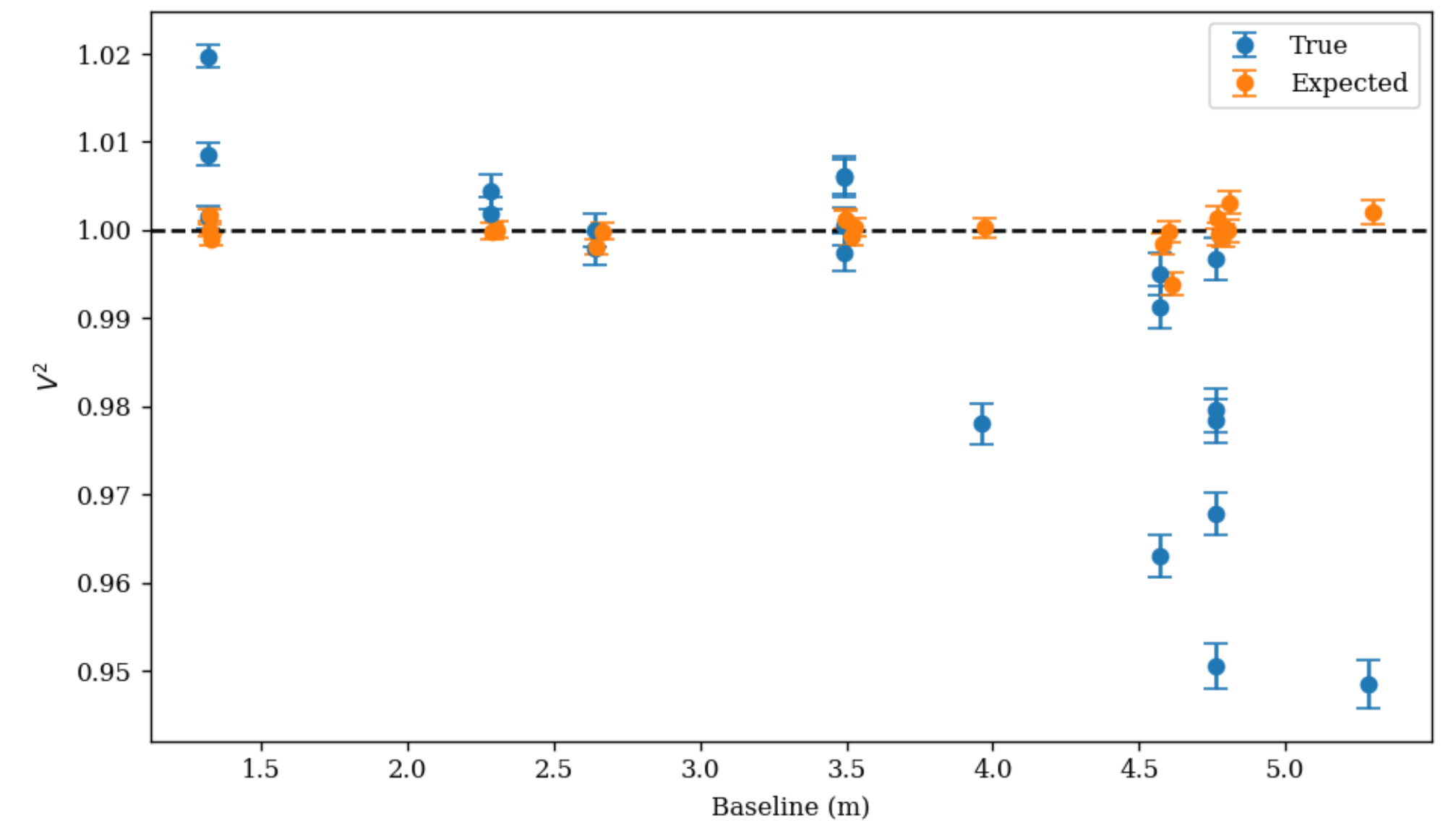
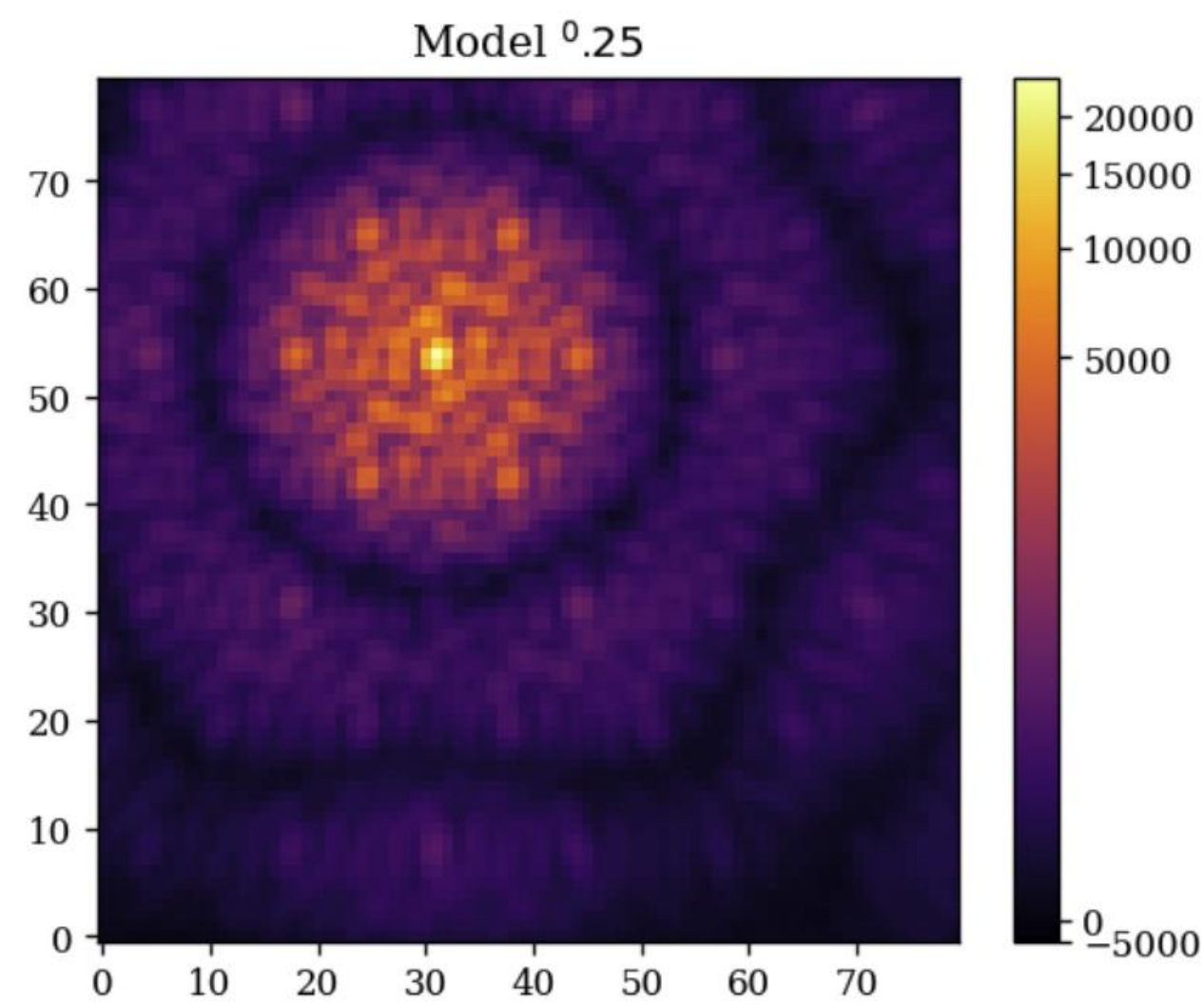
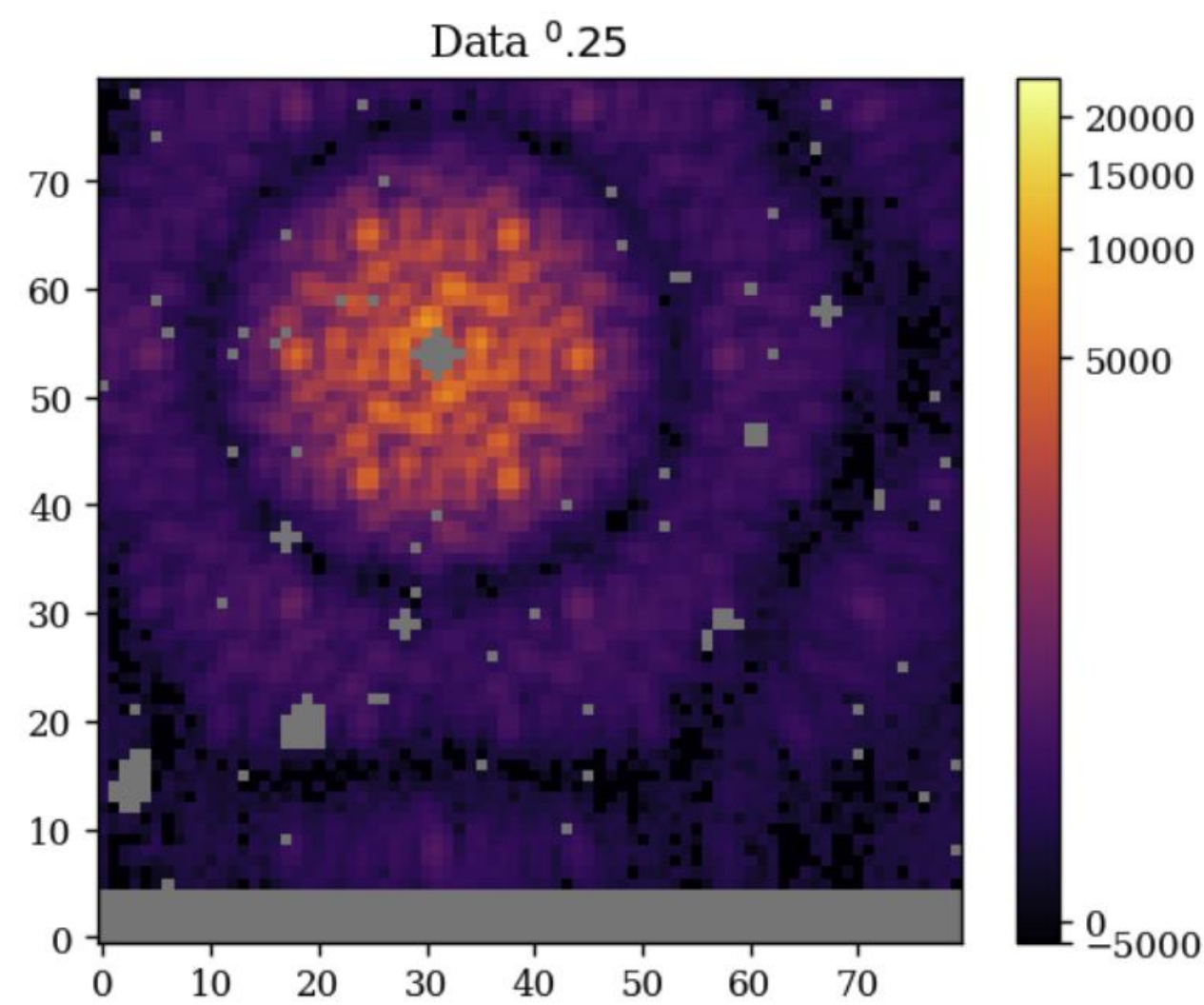
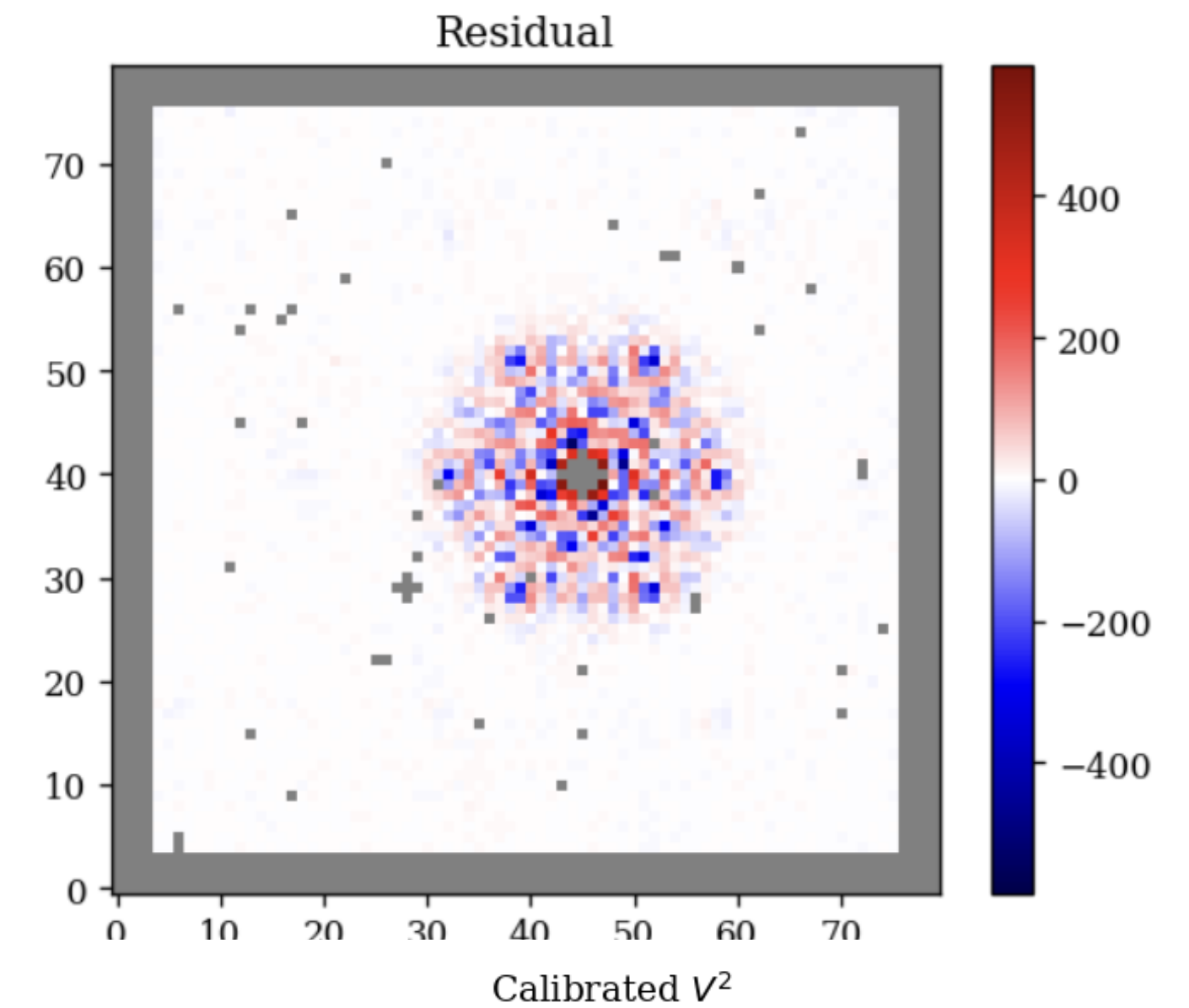
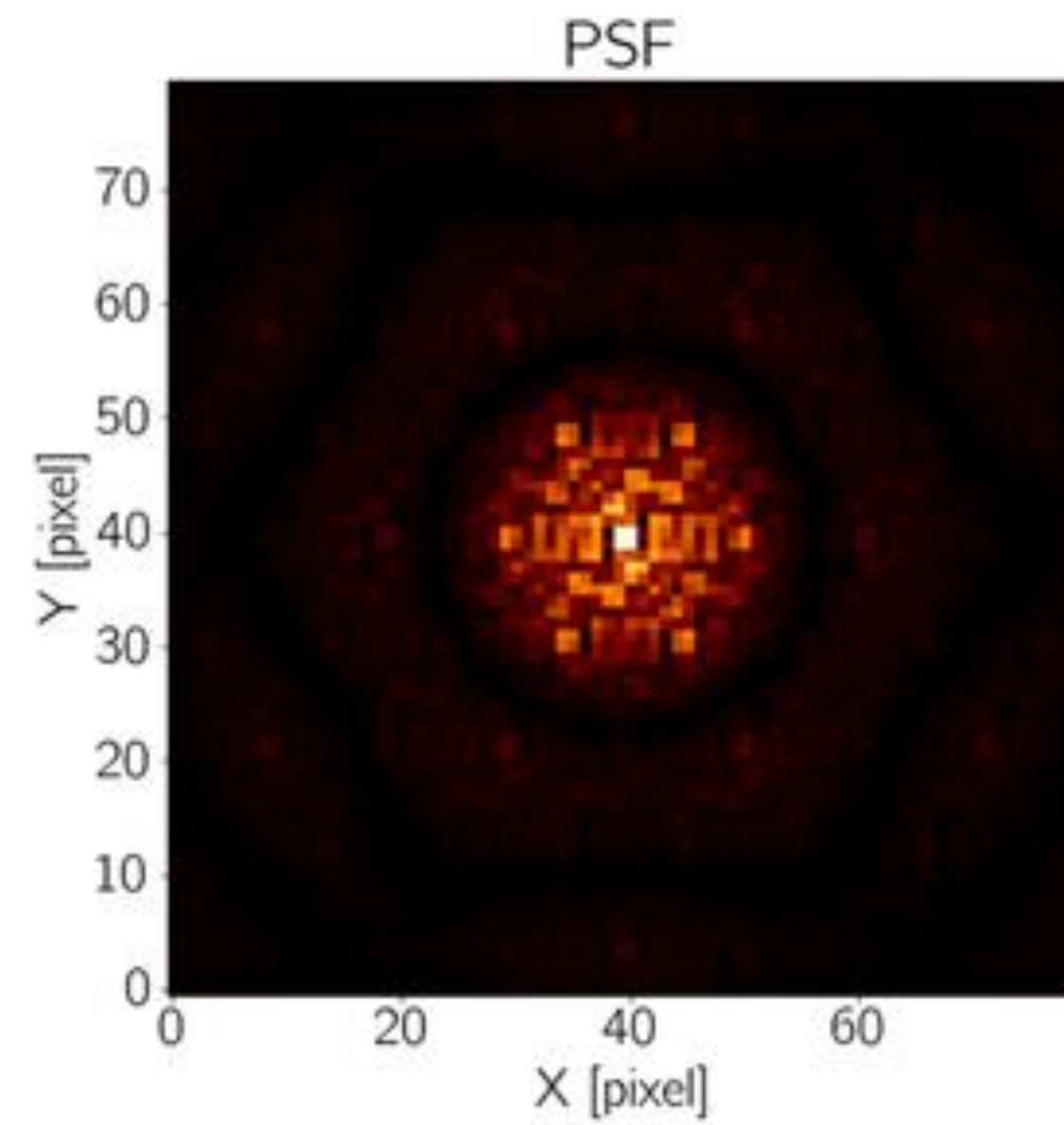
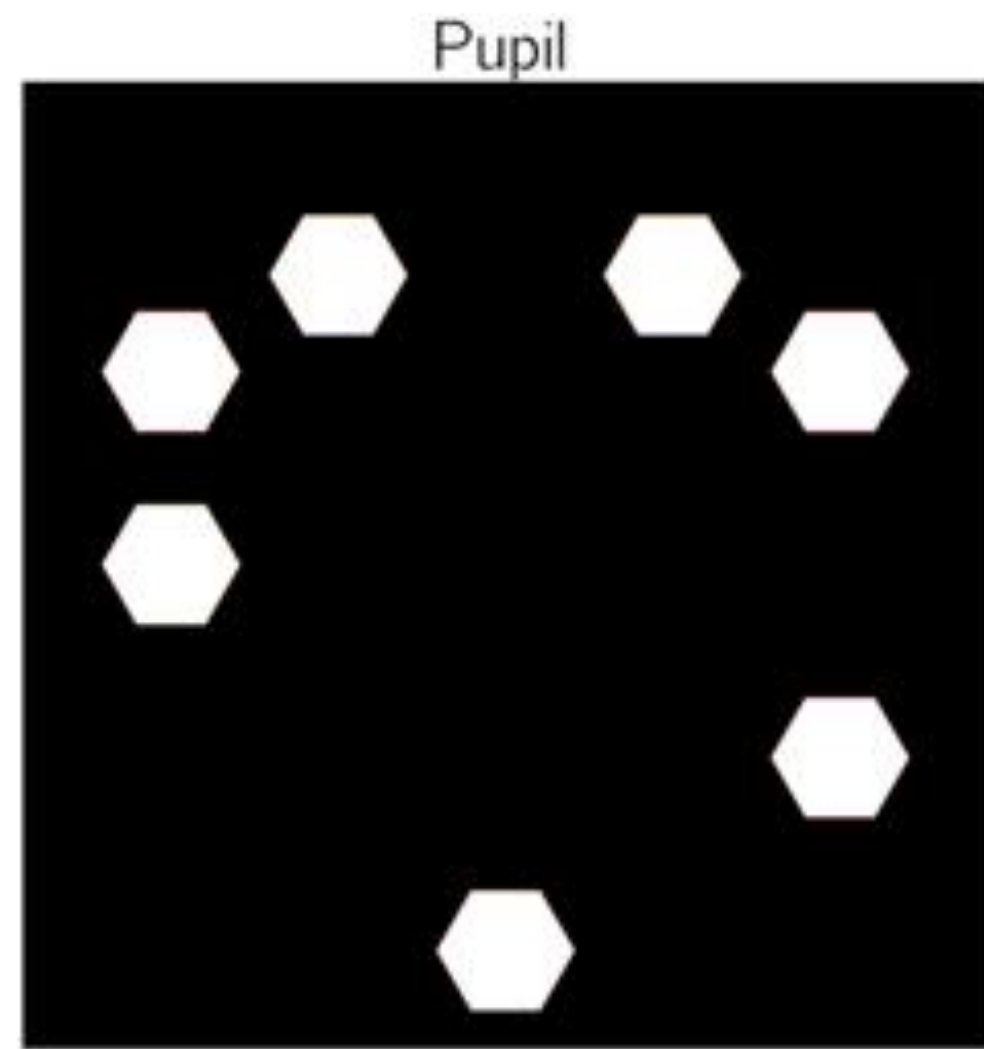
Louis Desdoigts



Ben Pope

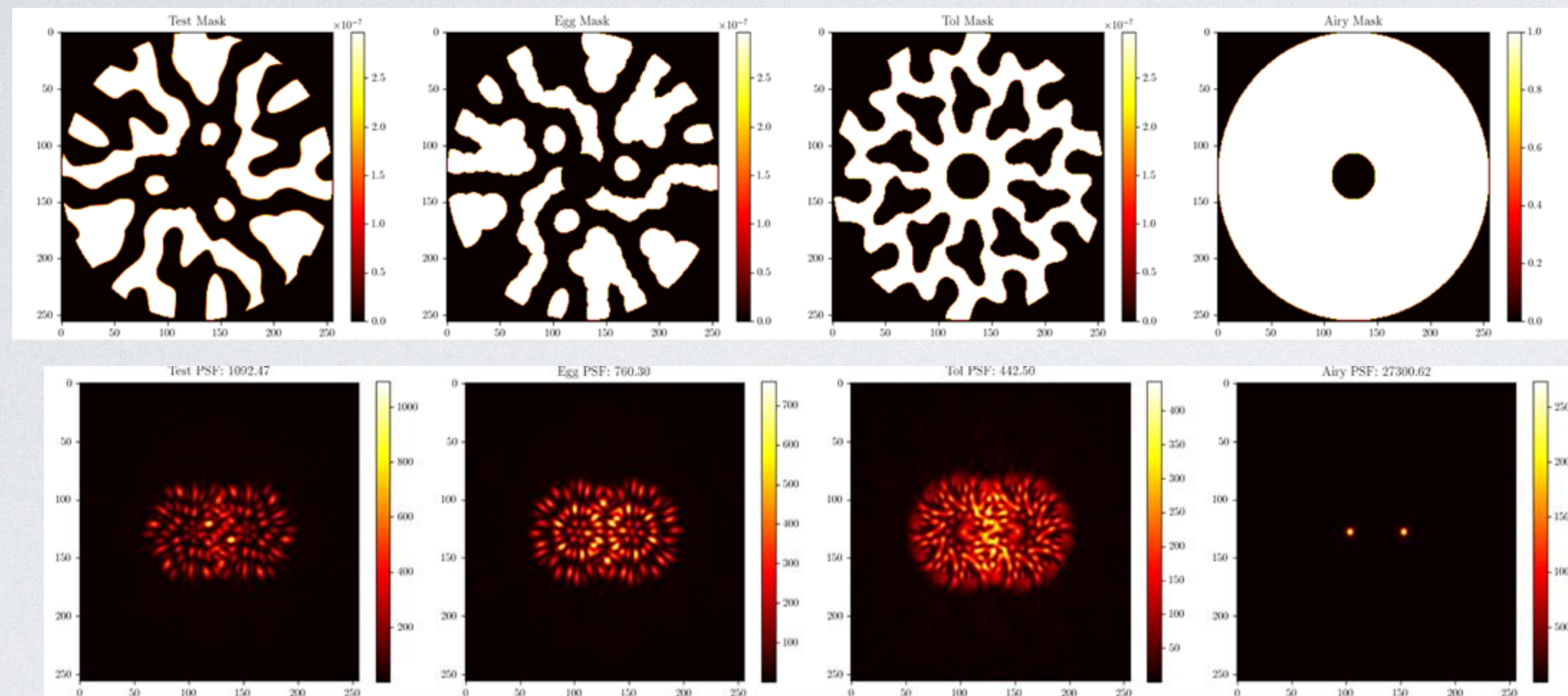


NIRISS AMI: THE JWST INTERFEROMETER

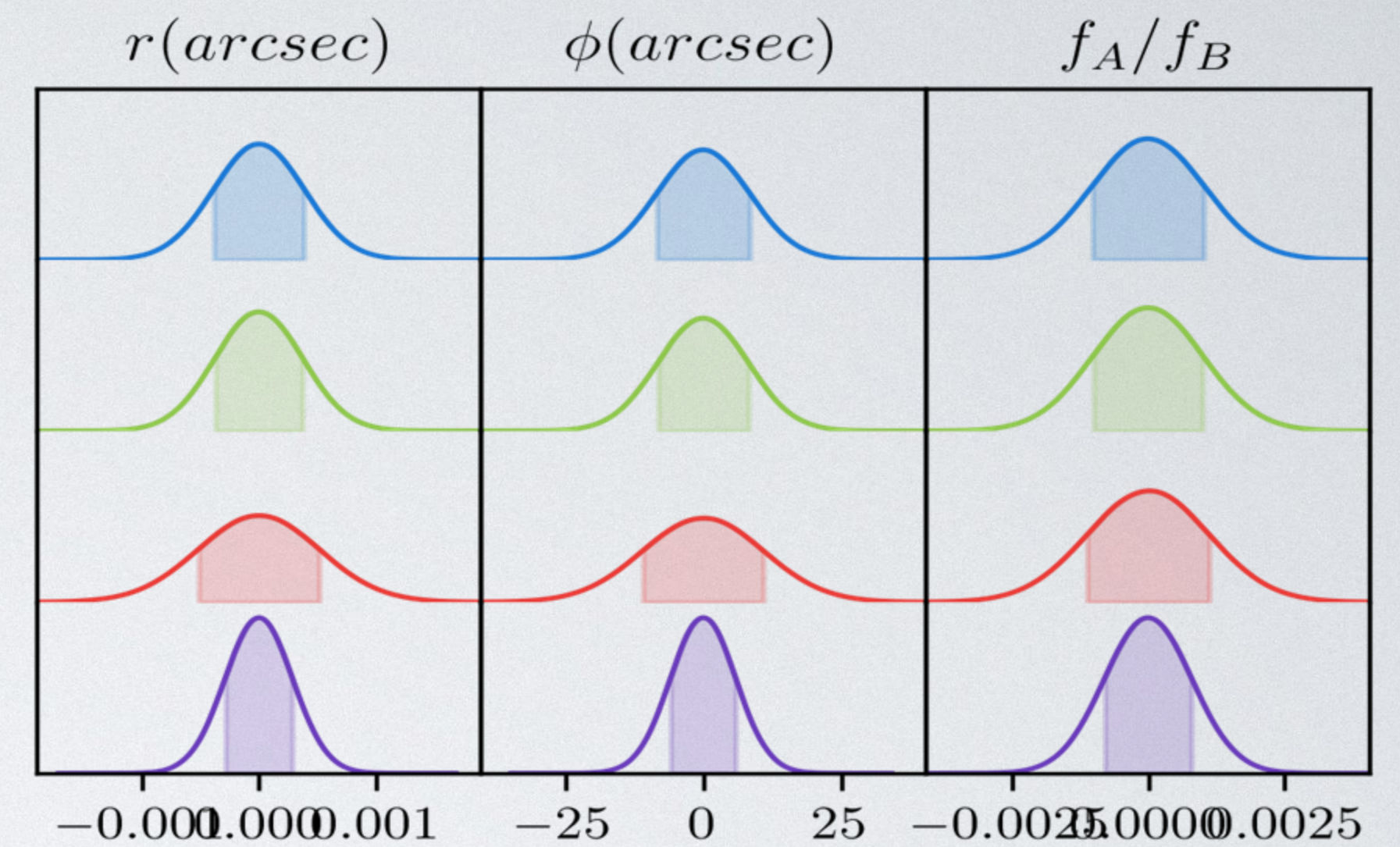




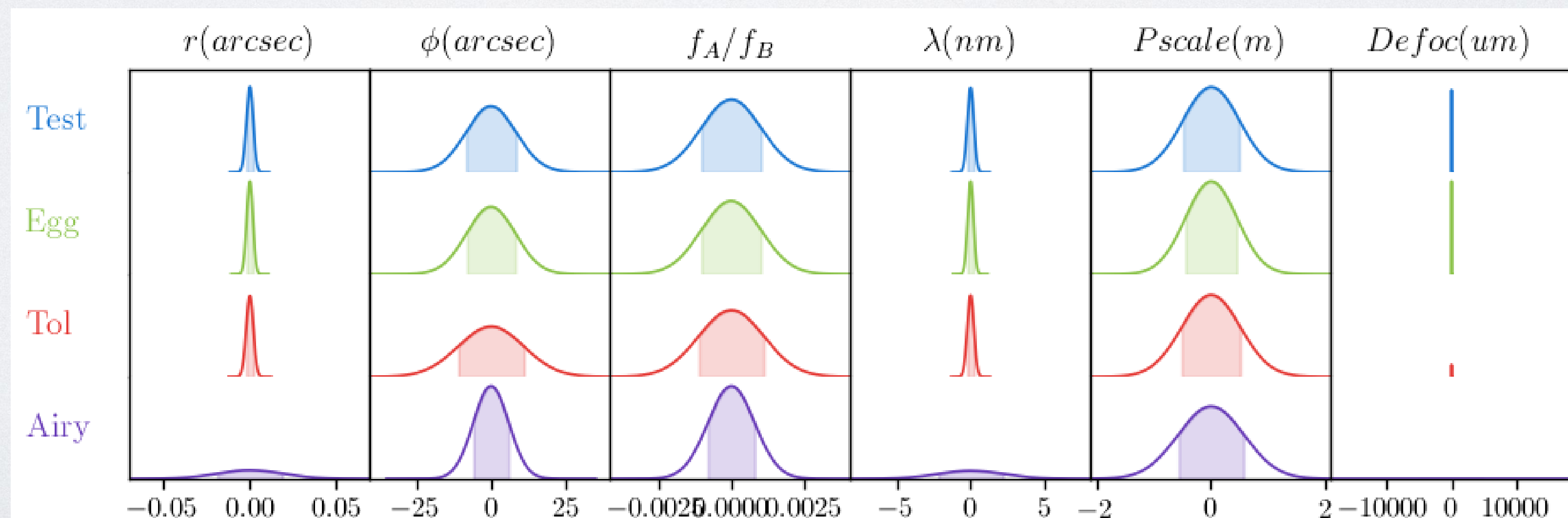
Posteriors: Separation, field angle, flux ratio

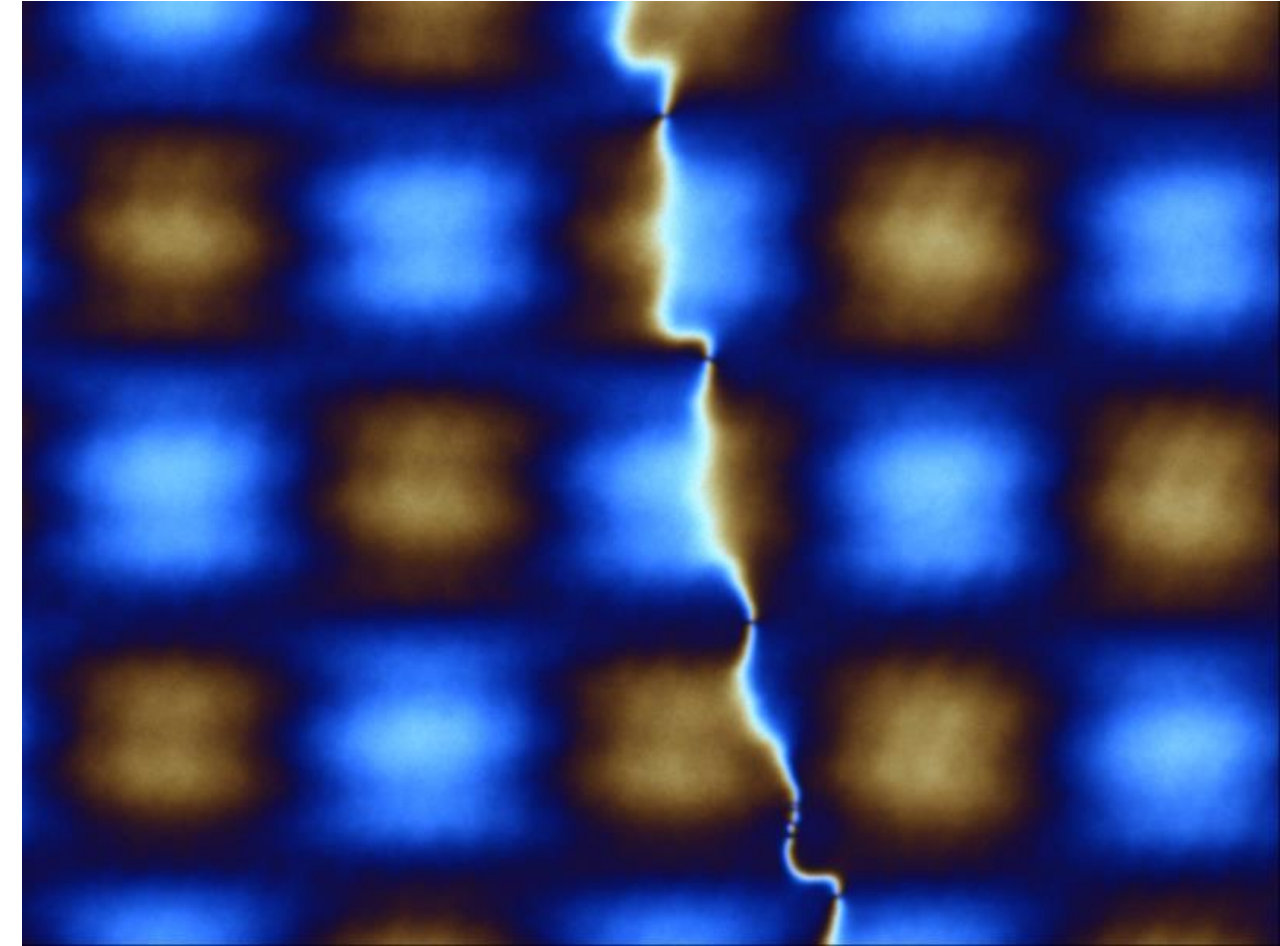
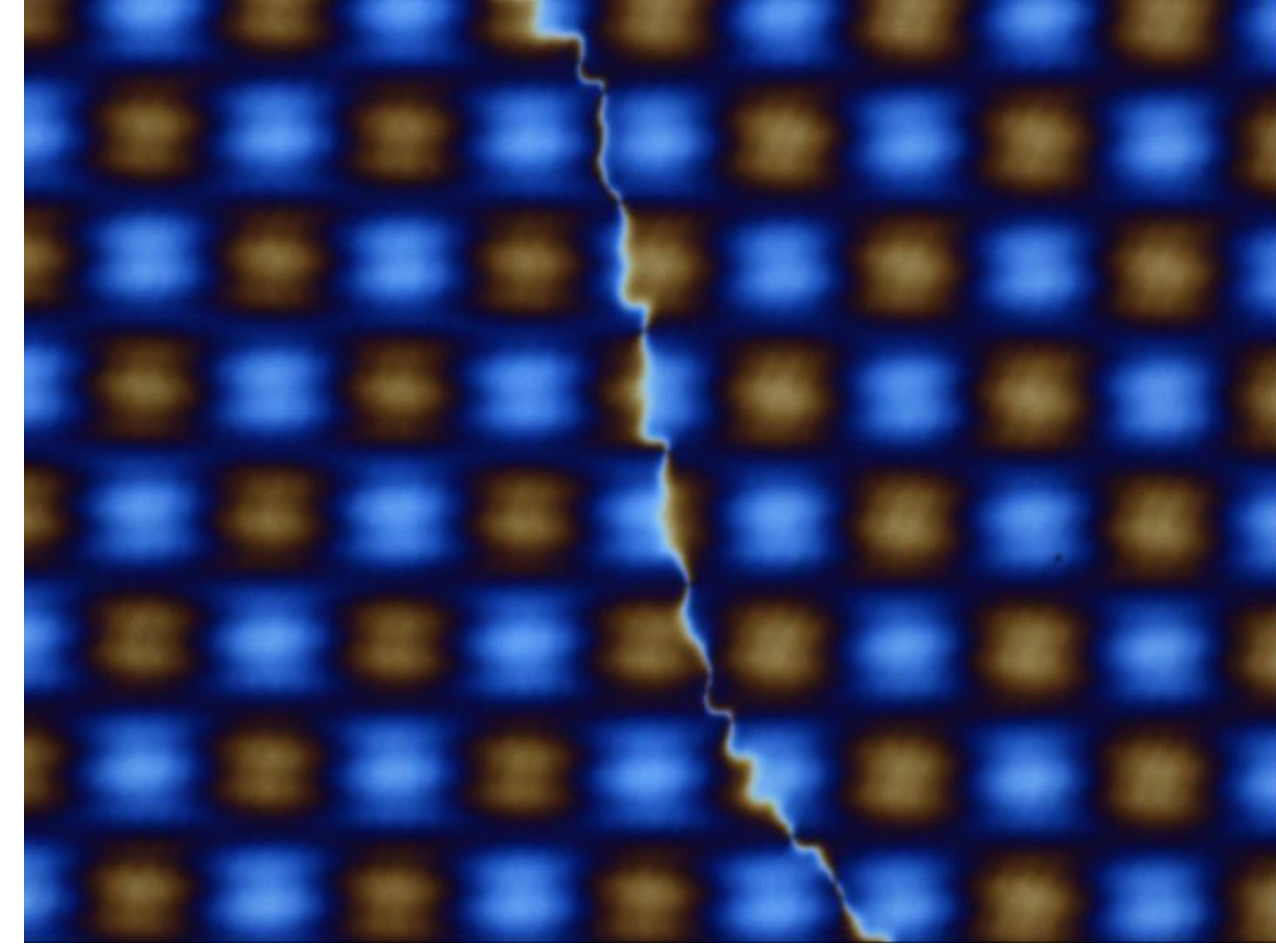
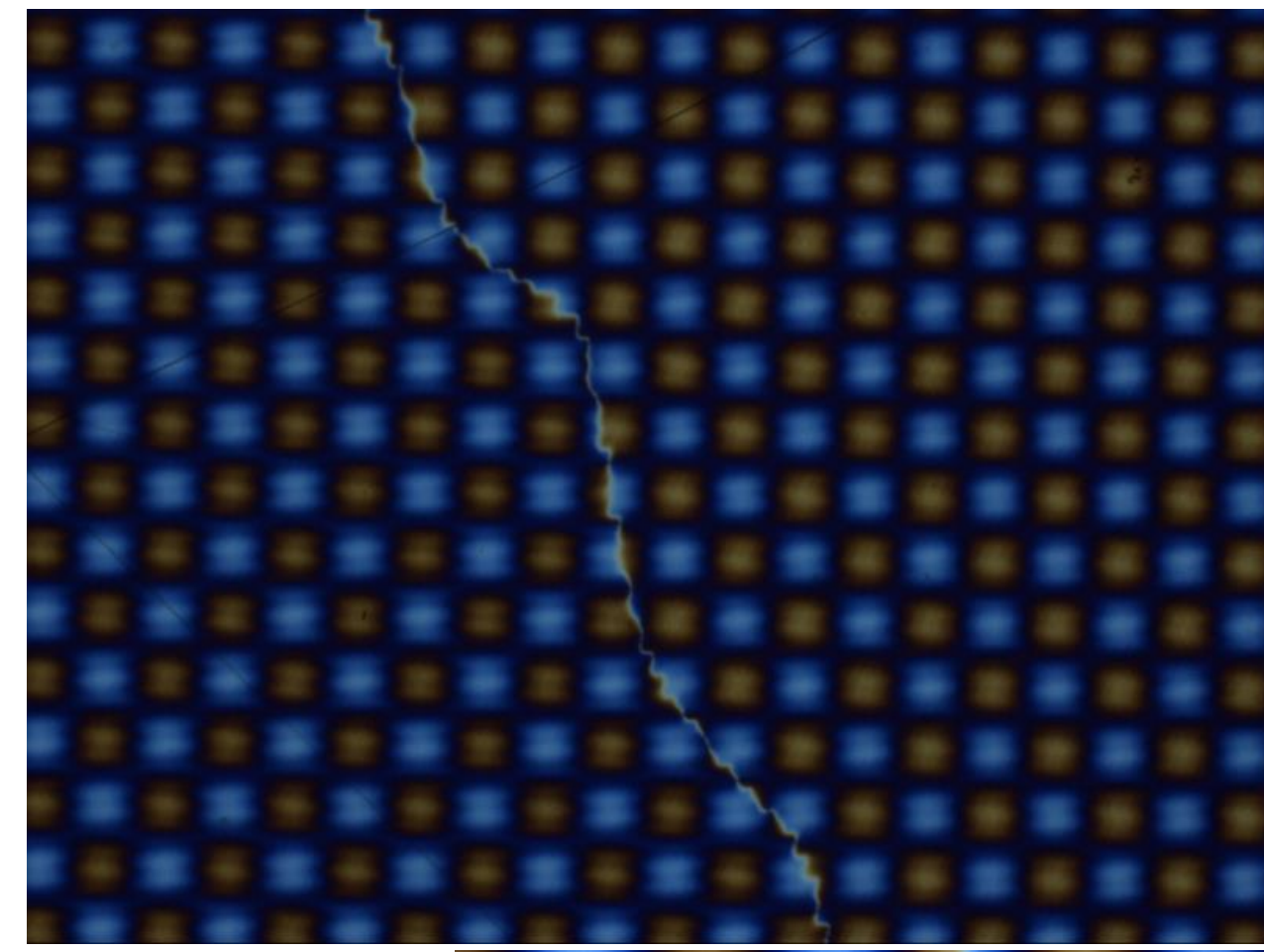
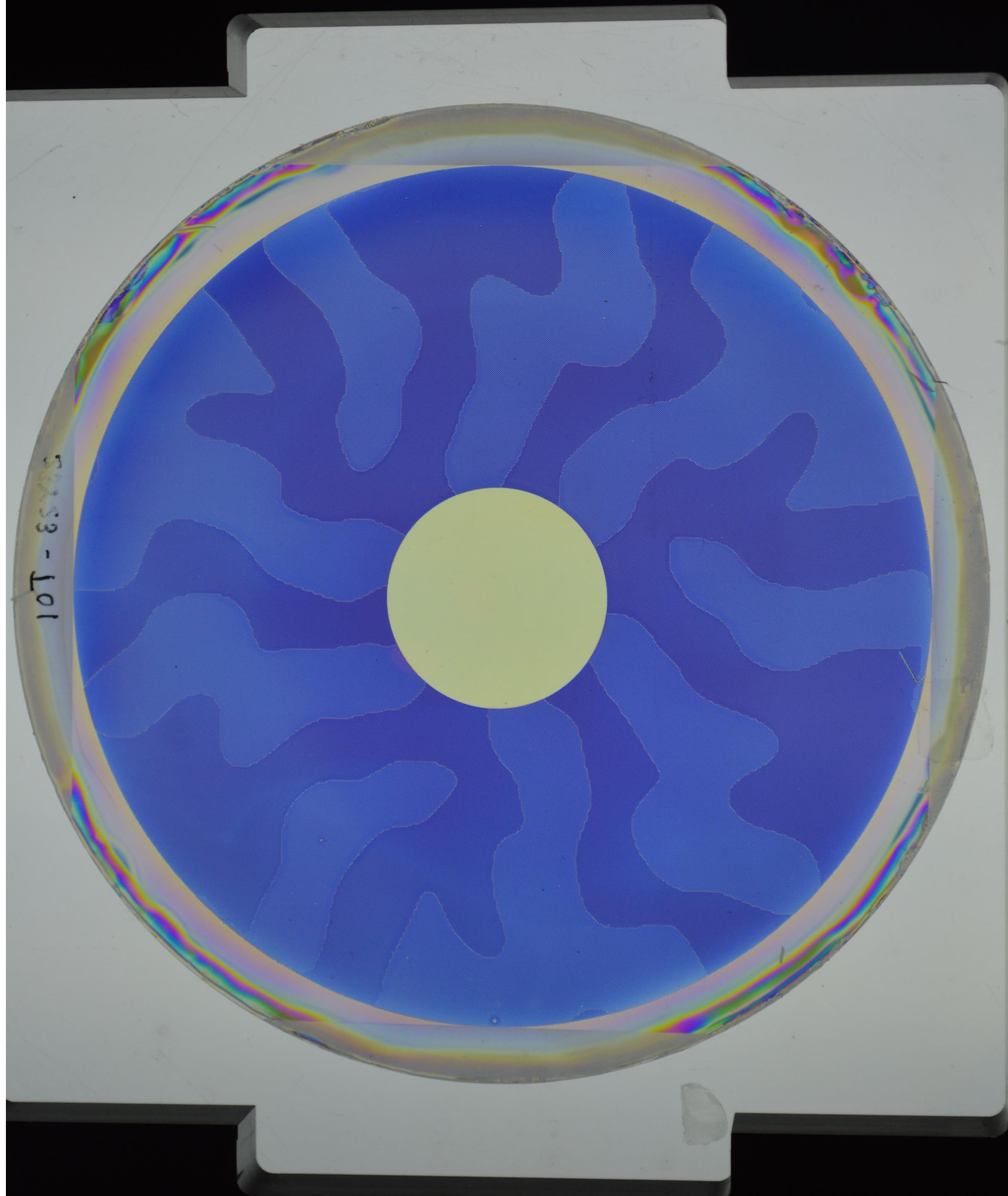


Test
 Egg
 Tol
 Airy



POSTERIORIS: ADDING WAVELENGTH, PLATE SCALE,

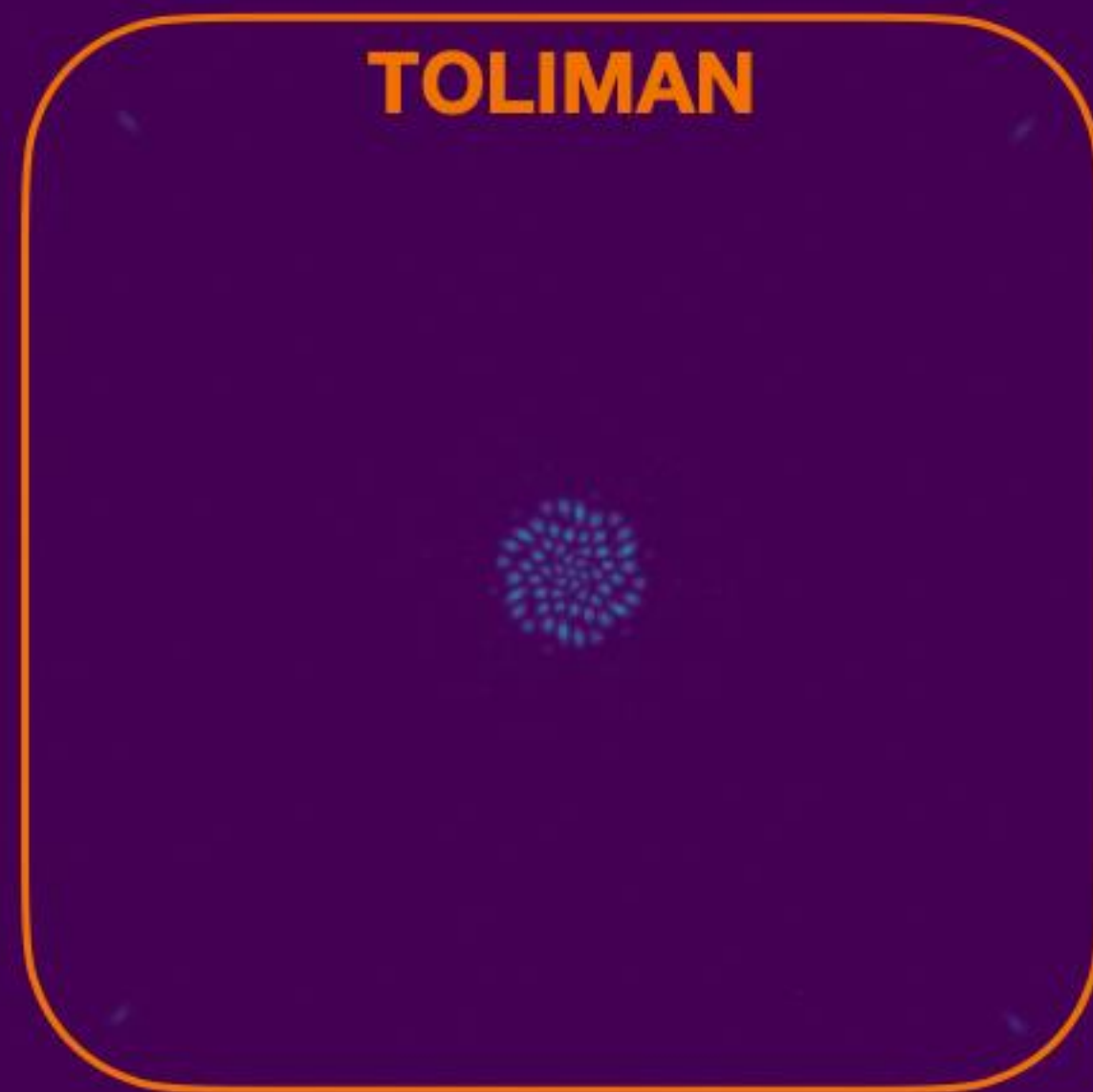




TOLIMAN PSF images

Linear PSF

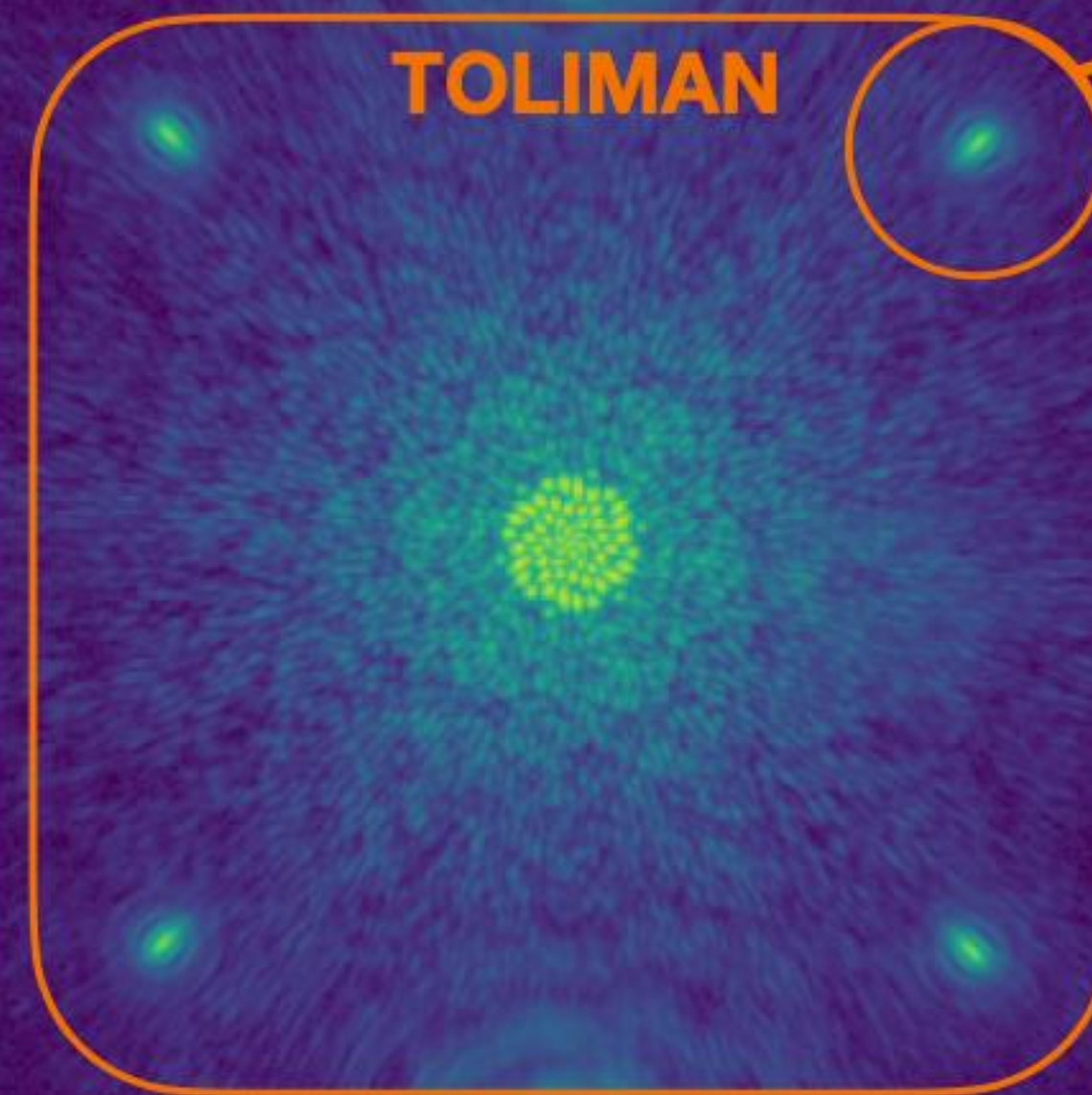
Grating mask



Grating mask

Log10 PSF

Grating mask



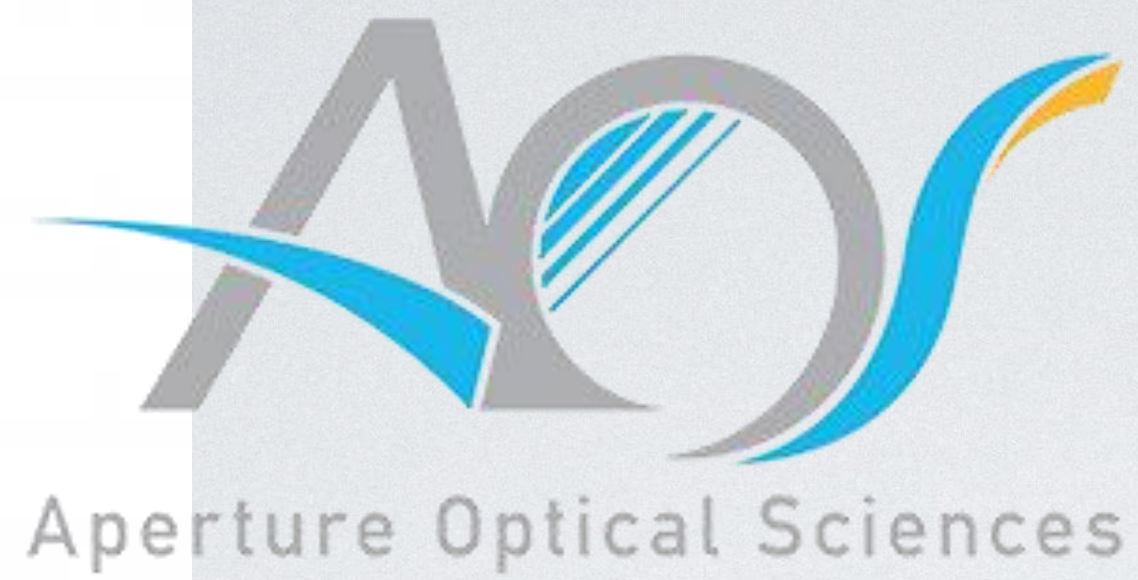
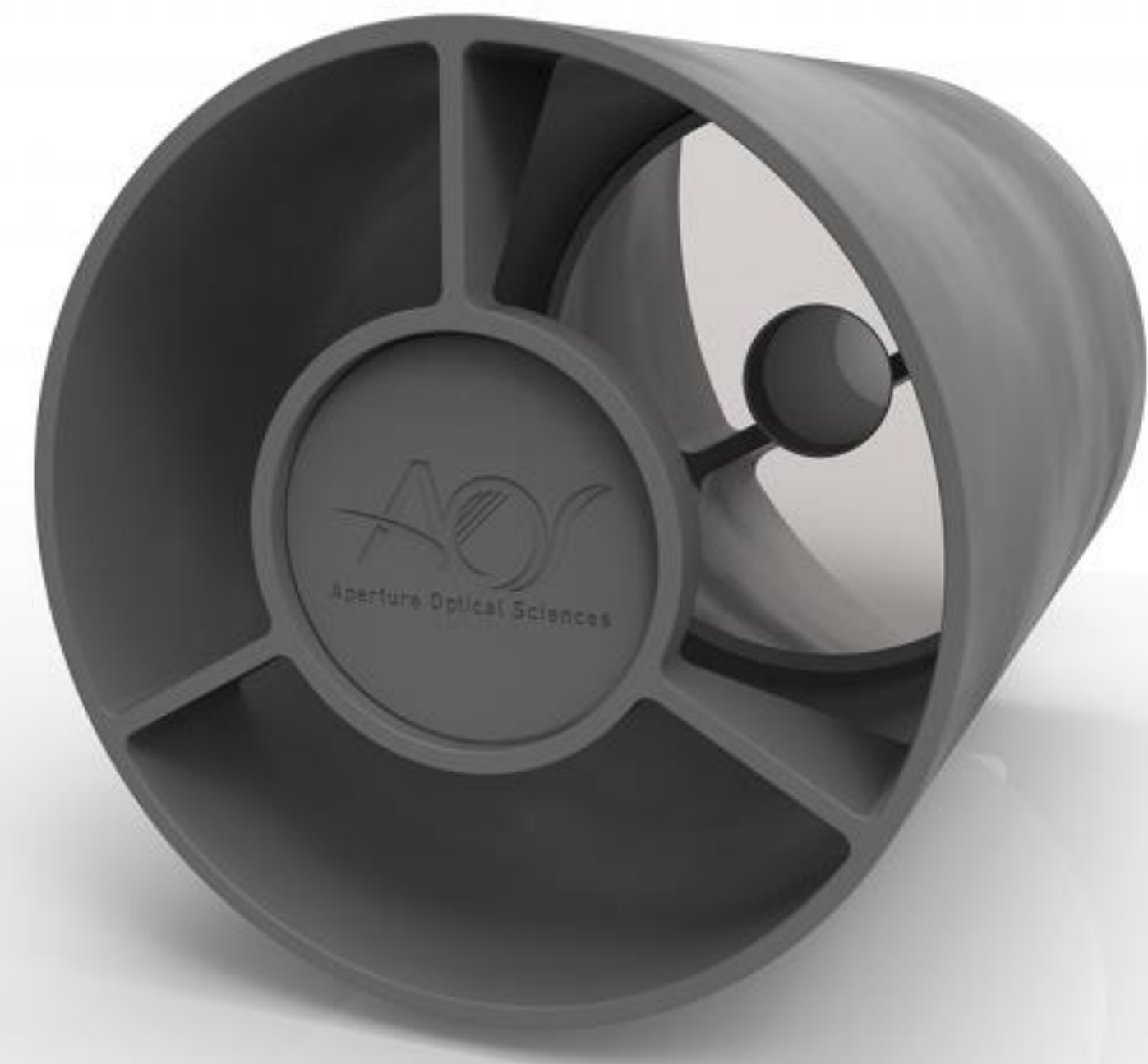
Grating mask

Diffraction spots have changed frequency to reduce PSF size. Also lower intensity

Linear PSF

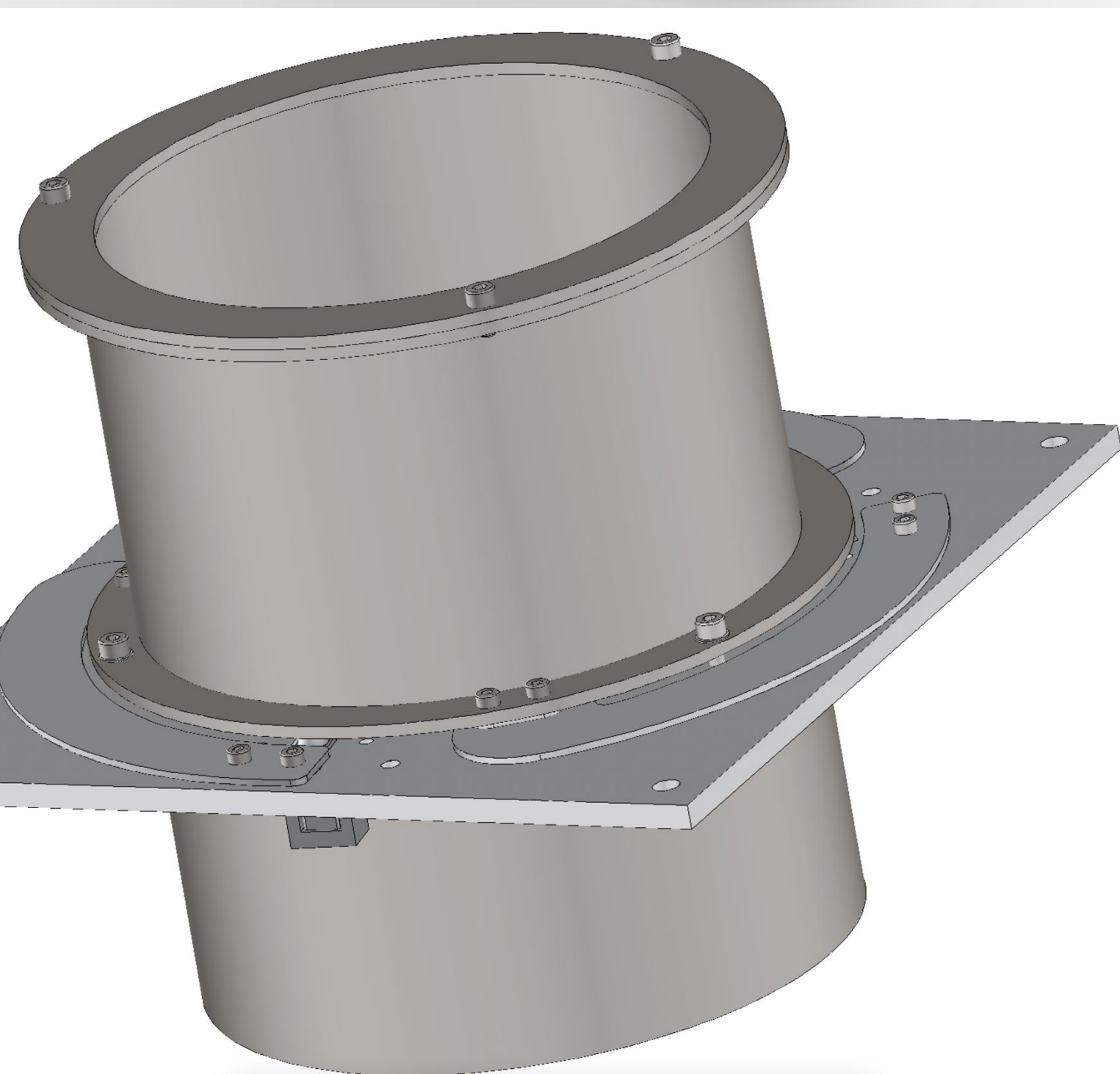
Log10 PSF





SiC Telescope

- Design has space heritage
- 130cm Primary Mirror
- F/10 custom prescription to match detector
- Pure Silicon Carbide structure
 - Low CTE, ideal for Toliman
- Deliver Q4 2024
- USYD developing mounting interface with piezo based tip-tilt system for fine pointing

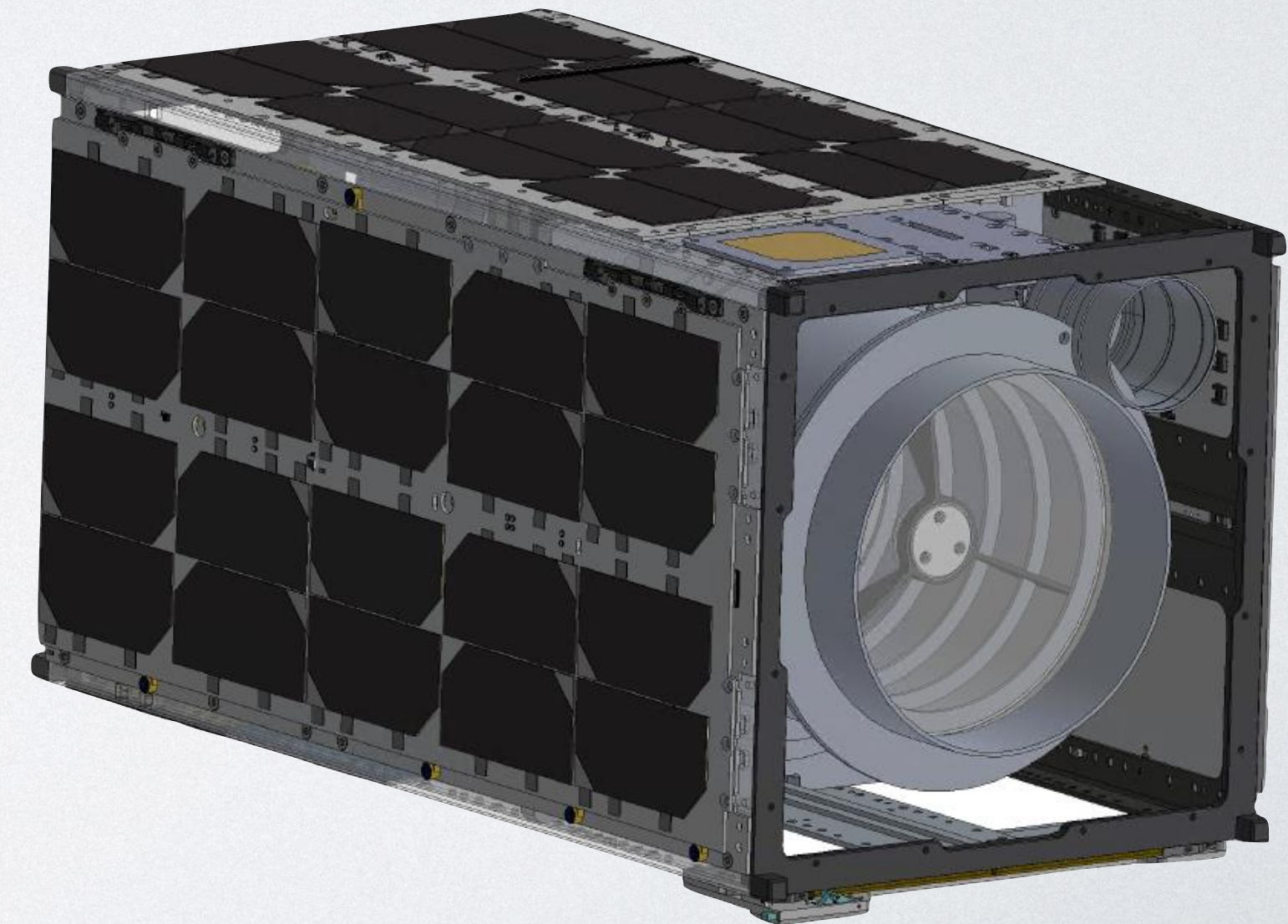
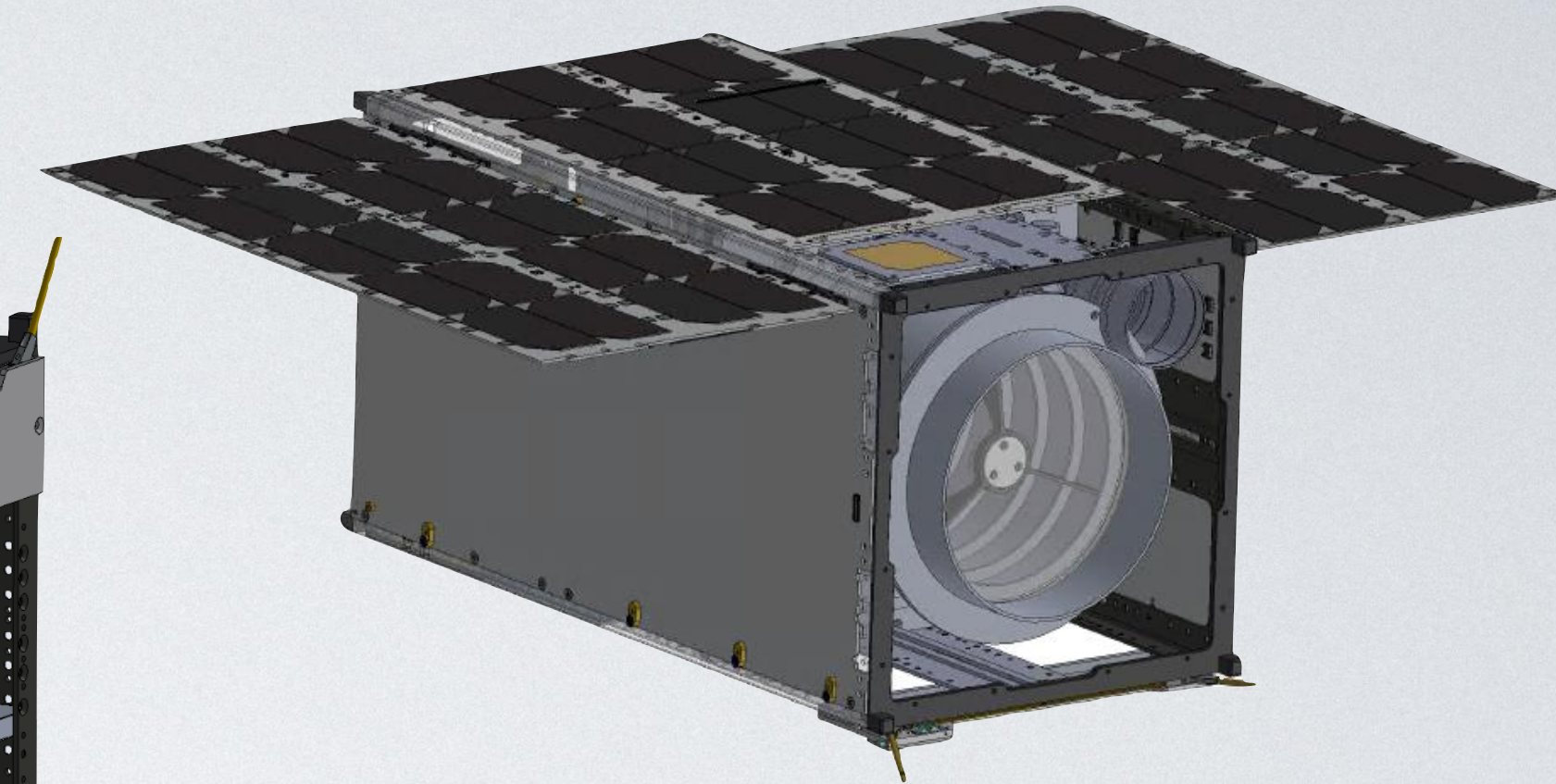
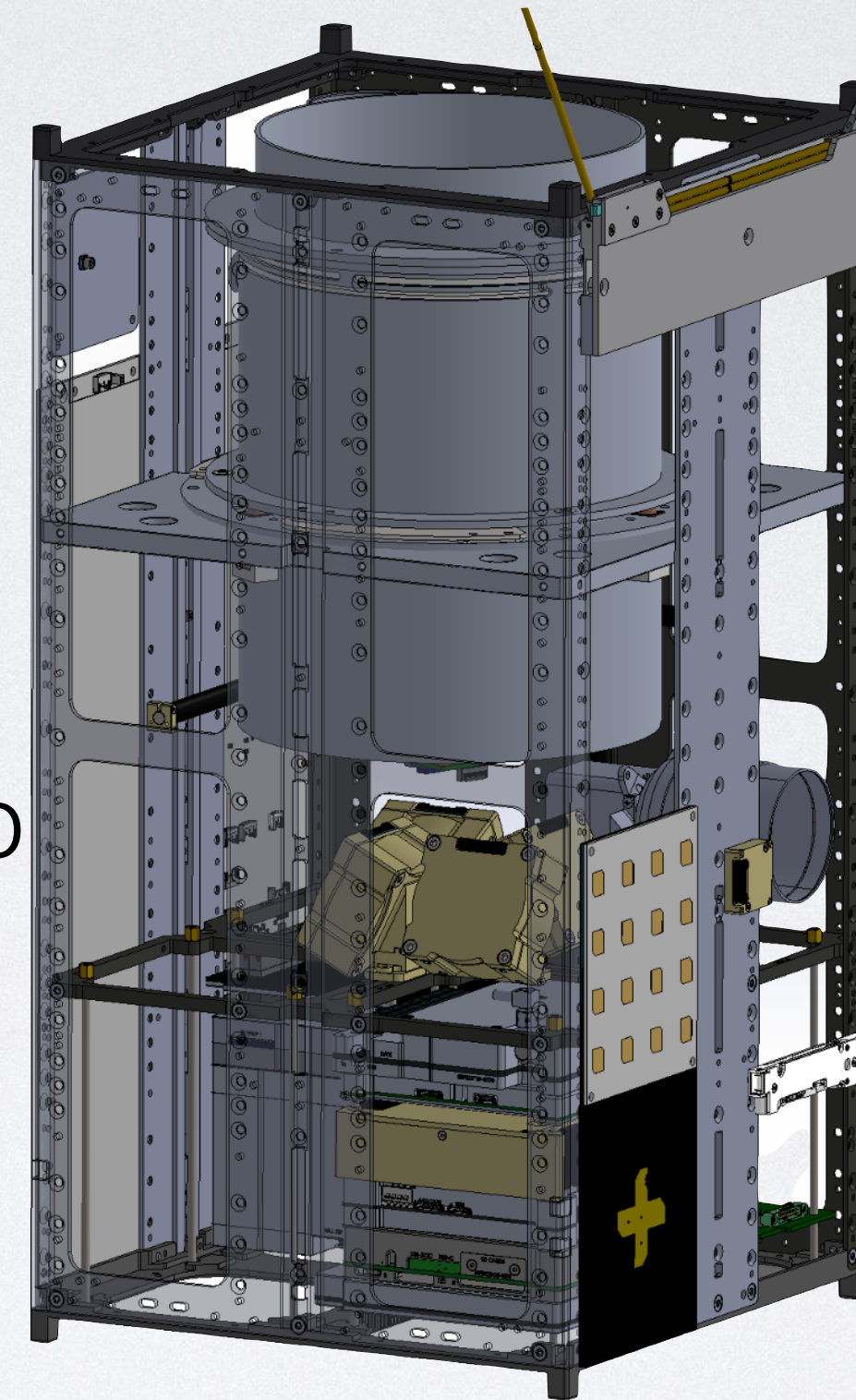




ENDUROSAT

16U Spacecraft

- 16U spacecraft
- X-Band downlink
- Plenty of power
- Advanced ADCS system*
- Full CubeSat as a service model, to be integrated with Saber systems
- Contract signed Dec 22
- Biggest challenges
- are pointing and thermal control





Mission: basic parameters & status

This program aims for Flagship ultra-high precision science outcomes on a CubeSat budget. Despite innovative design, specifications are demanding.

Major Mission Challenges (all in LEO):

- (1) Launch 2026, 16 U cubesat SSO orbit in 2-3 year mission
- (2) Target astrometry (narrow-angle) is <1 micro-arcsec
- (3) Simultaneous Photometry (10hz) and Spectroscopy ($R \sim 200$)
- (4) All hardware components fabricated or in process
- (5) Integration into Bus in Sofia 2025

Key Take Aways:

- Sydney team welcomes science/instrument collaborations. See me or Karel Valenta!
- Success is *at least* as dependent on principled reduction of data as on optics. Differentiable forward modelling with `dLux` has been a gamechanger (public on github).
- Unorthodox (coded-aperture) imaging systems might have reach / application to YOUR mission (Talk to me – have ideas!)

University of Sydney TOLIMAN team



Peter Tuthill



Chris Betters



Connor Langford



Karel Valenta



Fred Crous



Louis Desdoigts



Max Charles

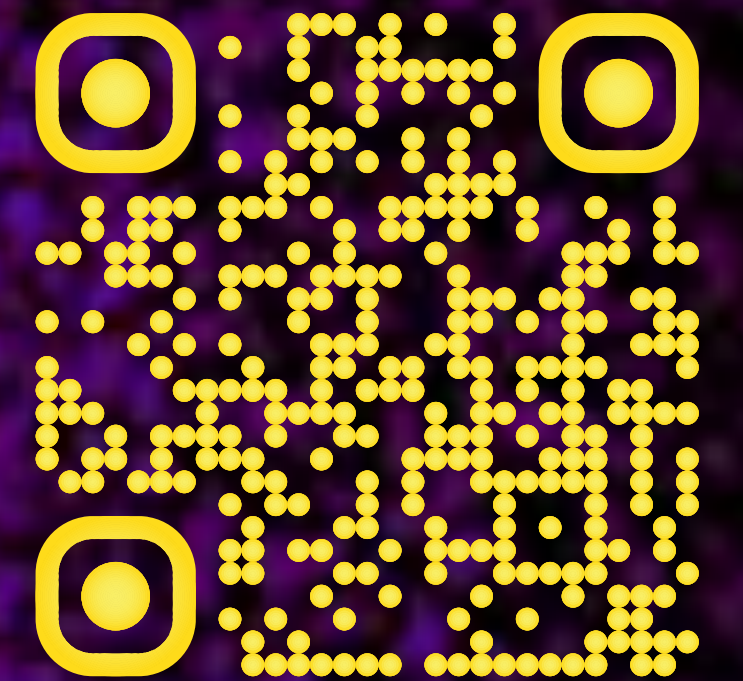


Grace Piroscia

TOLIMAN



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